

DAMPE: a particle detector in space

Giovanni Ambrosi

INFN Perugia, Italy

(on behalf of DAMPE collaboration)



CERN Detector Seminar
July 15°, Geneva

The physics goals

High energy particle detection in space

- Study of the cosmic electron and photon spectra
- Study of cosmic ray protons and nuclei: spectrum and composition
- High energy gamma ray astronomy
- Search for dark matter signatures in lepton spectra

Detection of
2 GeV - 10 TeV e/ γ
50 GeV - 500 TeV protons and nuclei
with excellent energy resolution , tracking precision
and particle identification capabilities

- Exotica and “unexpected” , e.g. GW e.m. counterpart in the FoV

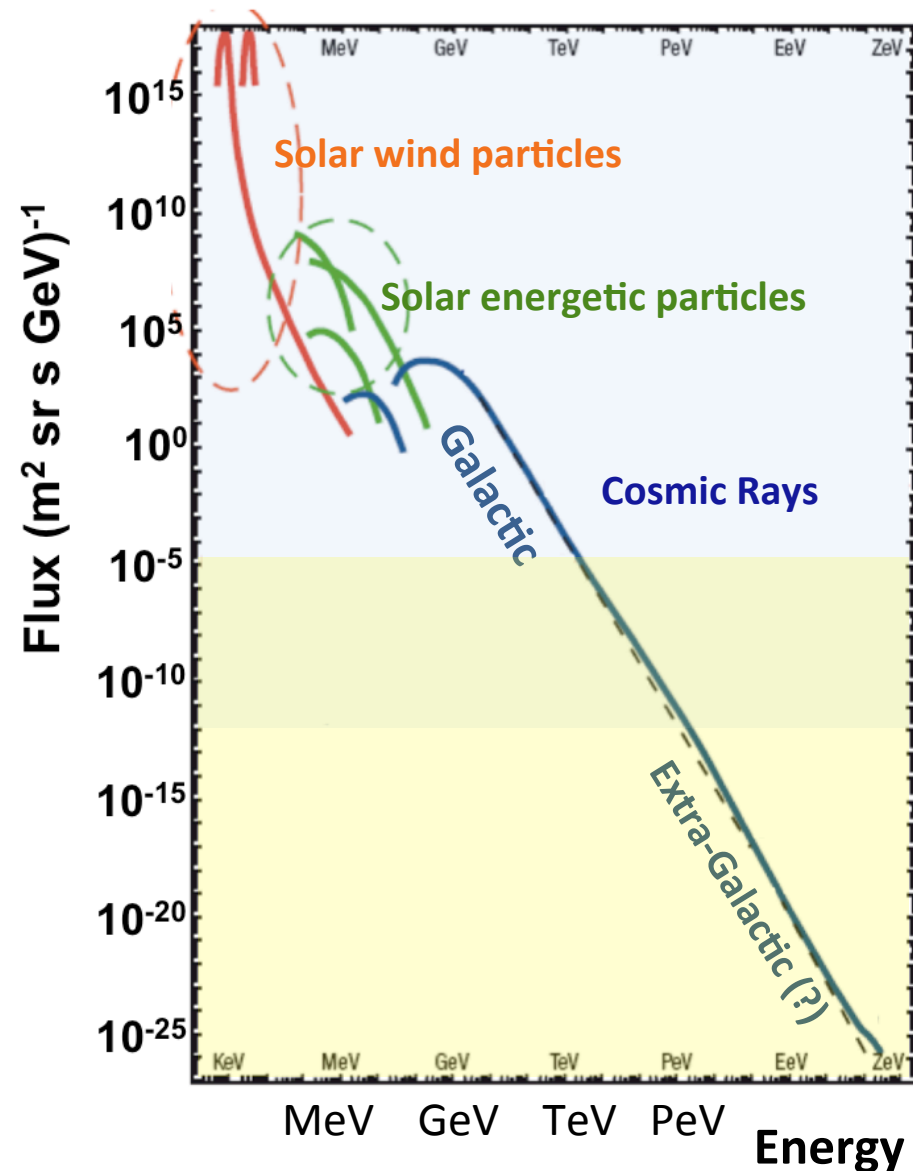
The CR spectrum: the overall picture

Direct measurements:

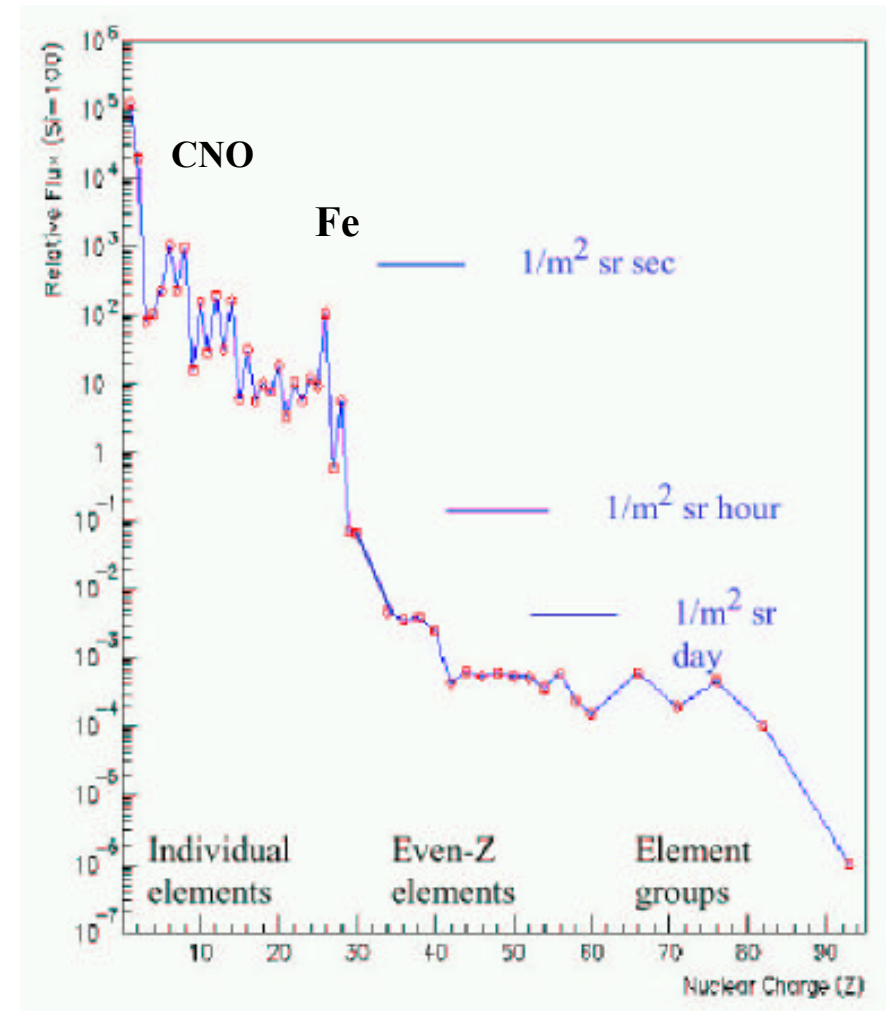
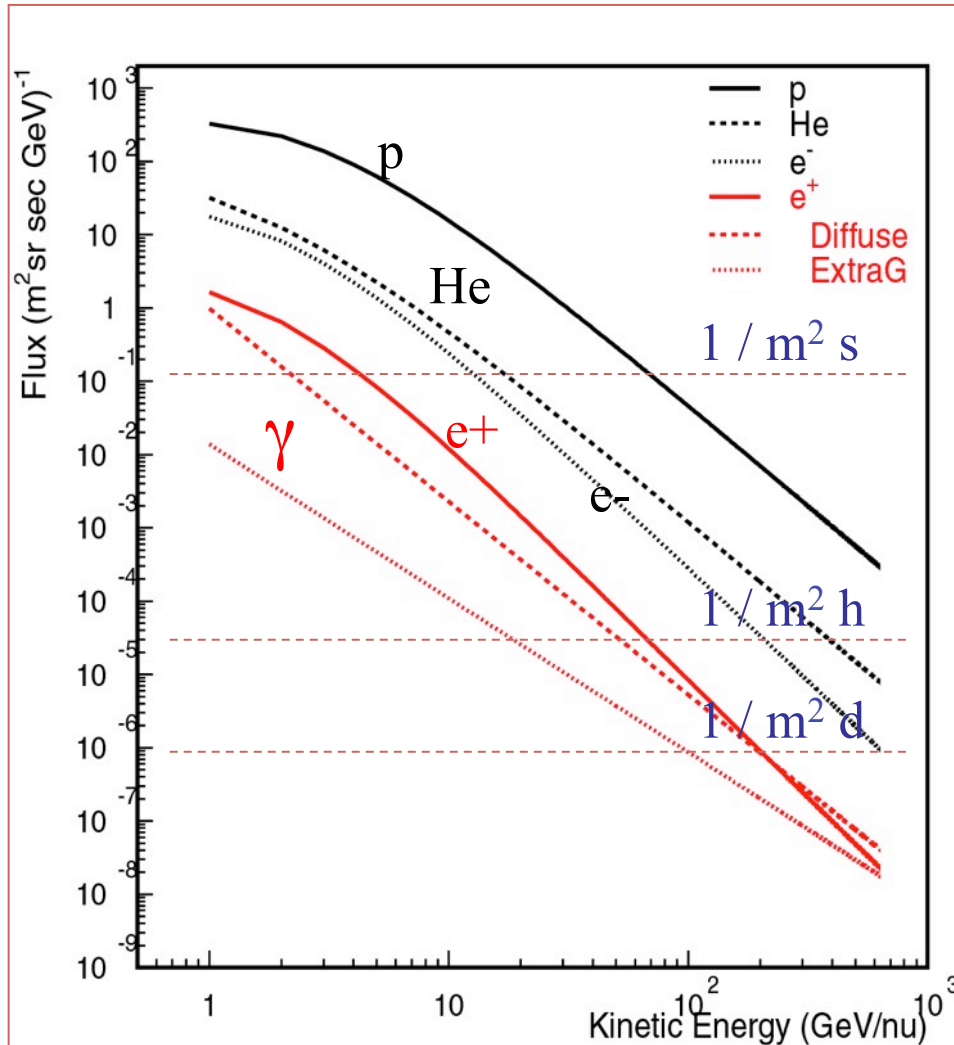
- ☺ Particle identification/Energy calibration, anti-matter
- ☹ Space: Weight/Size constraints limit the energy range (< PeV)

Indirect measurements:

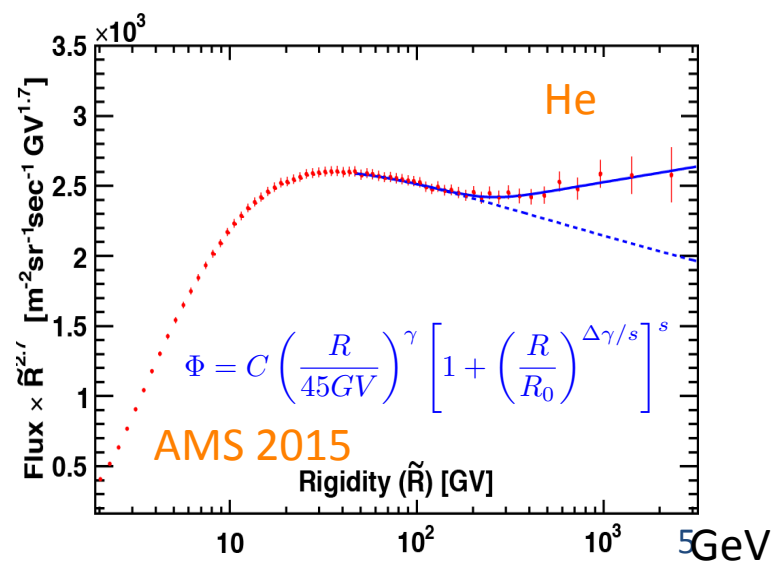
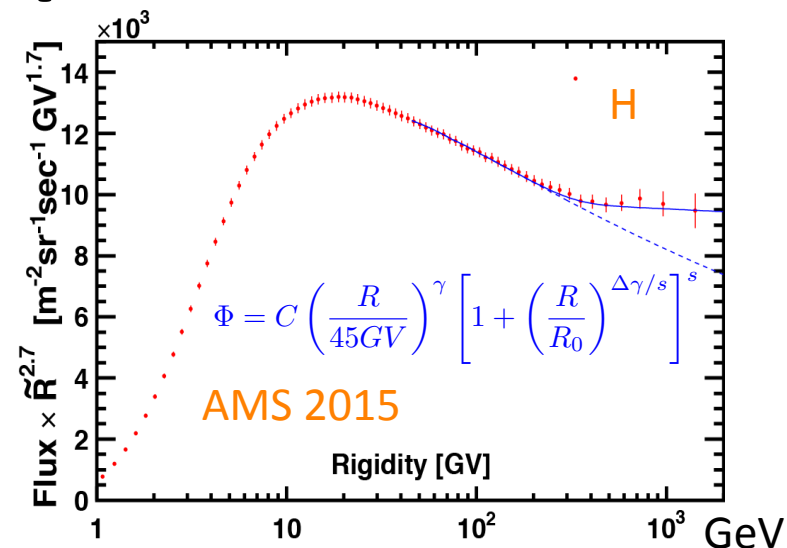
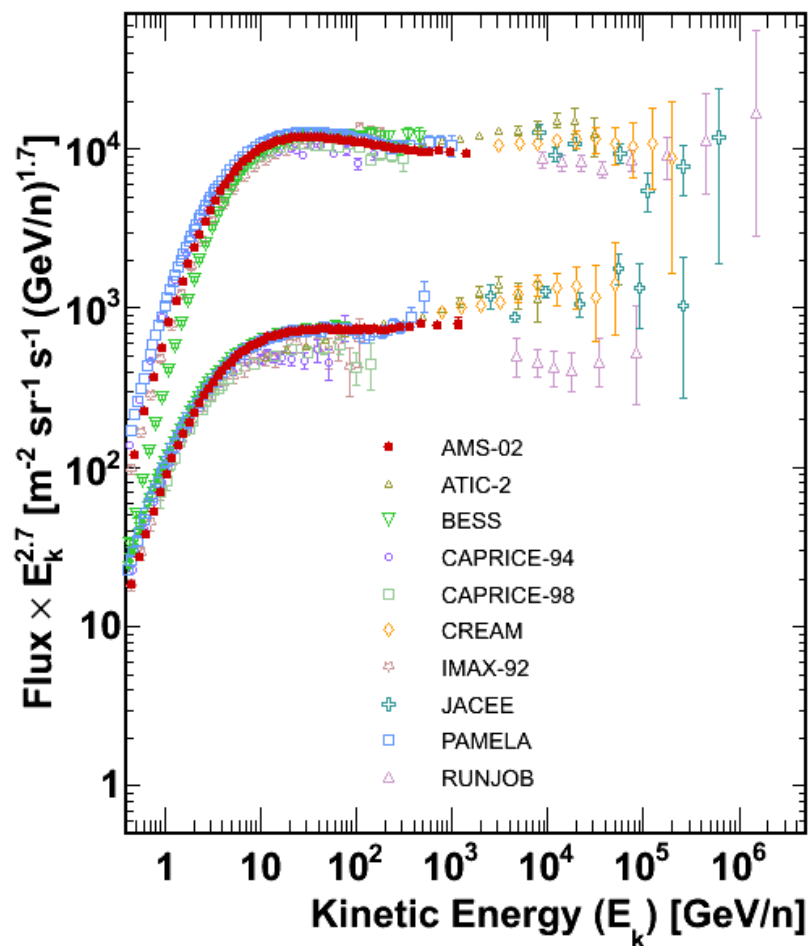
- ☺ Ground: Extended energy range (>PeV)
- ☹ Pid/Energy : dependence on modelling of atmospheric interactions



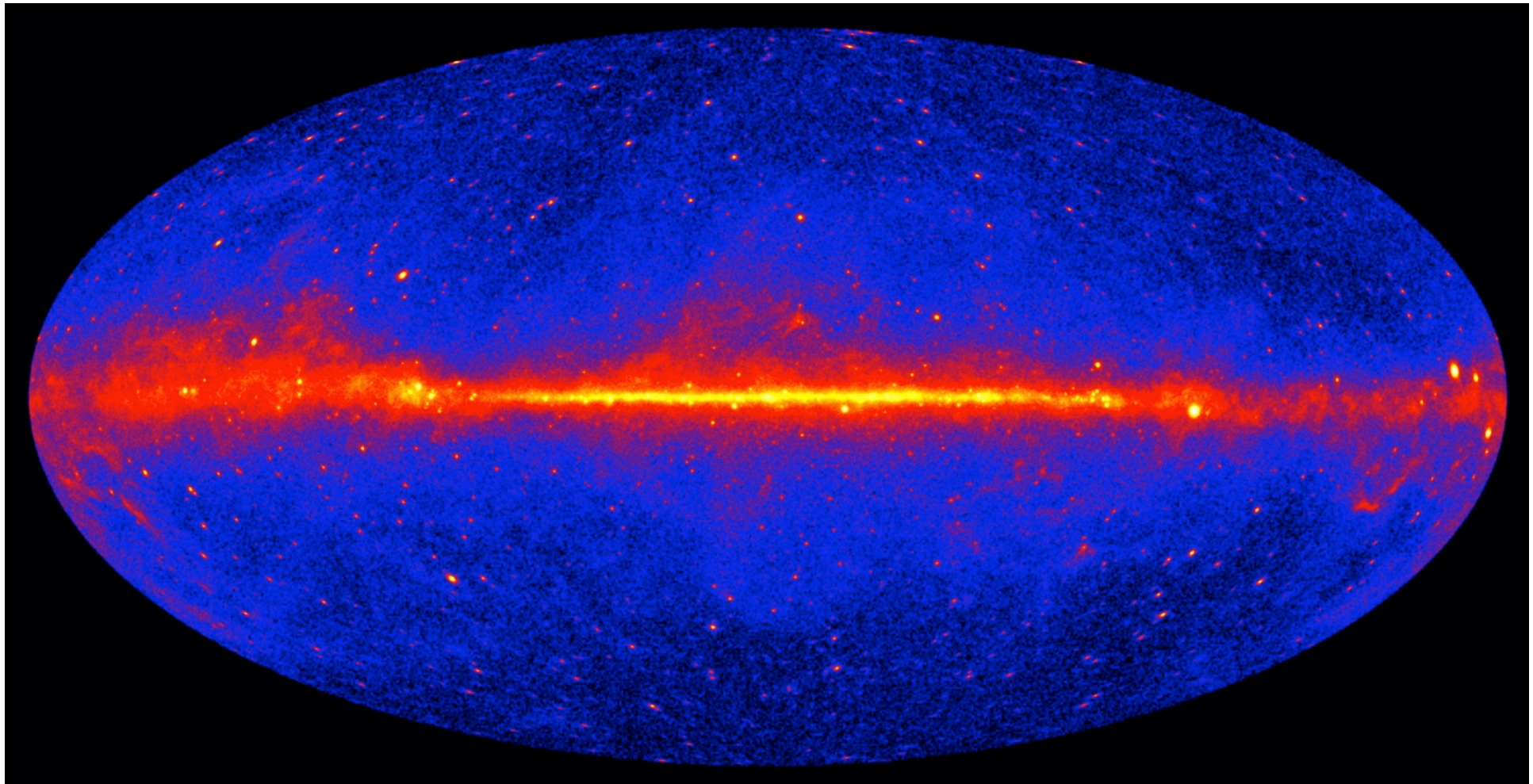
CR flux and composition



AMS-02 : the smooth change of spectral index



FERMI all sky map





DON'T LET THE BRIGHT
LIGHTS FOOL YOU

THE DARK SIDE

CONTROLS THE UNIVERSE

OUR UNIVERSE

STARS: 0.5%

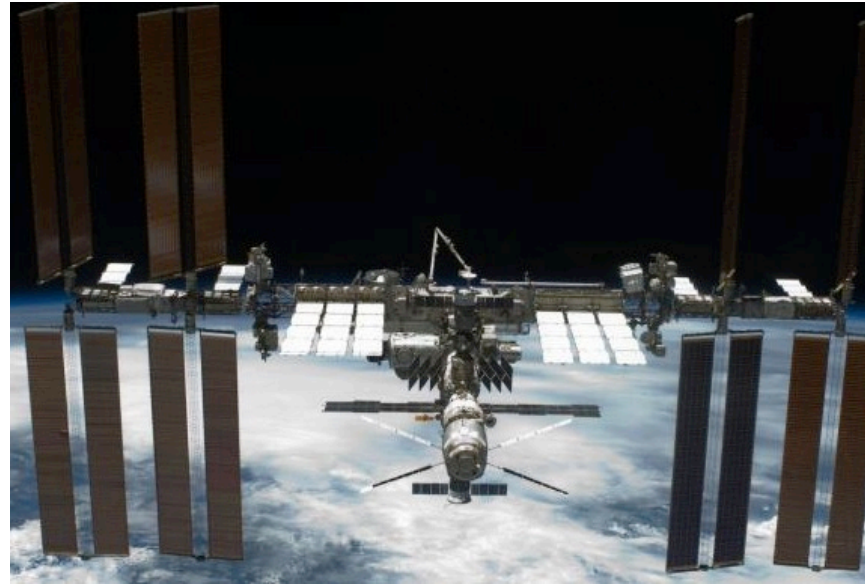
DARK
MATTER: 33%

DARK
ENERGY: 66%

DARK MATTER HOLDS IT TOGETHER

DARK ENERGY DETERMINES HIS DESTINY

The experimental challenge



DIRECT \neq EASY

No atmosphere:

Stratospheric Balloons
Space

Limits on size and time:

Detector design focused on specific
measurements

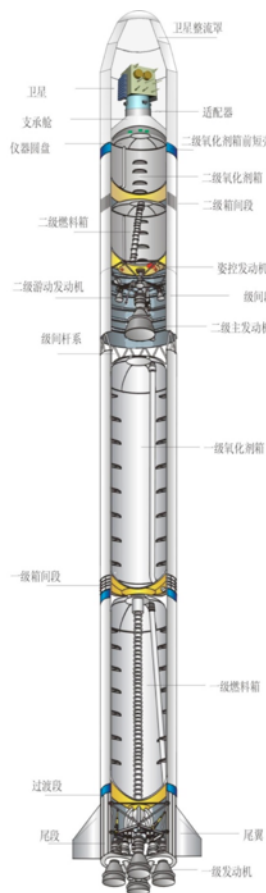


The instrument we need has ...

- performance a la 'particle physics':
 - high resolution measurements of energy, charge, particle id, momentum, velocity
- characteristics to properly work in the space environment:
 - Vibration (9 G rms) and acceleration (up to 500 G)
 - Temperature variation (day/night $\Delta T = 100^{\circ}\text{C}$)
 - Vacuum (10^{-10} Torr)
 - Orbital debris and micrometeorites
 - Radiation (Single Event Effect)
- limitation in weight, power, bandwidth and maintenance
- compliant with EMI/EMC specs
- Produce two pieces: Qualification Model and Flight Model

exact stress and limitation values depend from the detail of the mission

Operation in space

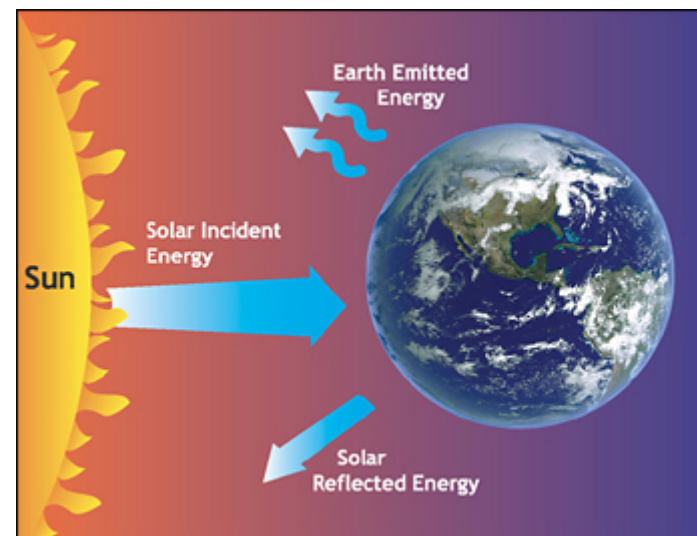


Mechanical stress at launch:

- Static acceleration
- Random vibration
- Sinusoidal vibration
- Pyroshock

Life in space:

- Thermal stresses due to Sun-light (seasonal / day-night effects)
- Vacuum



Careful Design, Model validation and Qualification are needed to ensure *highest possible reliability*

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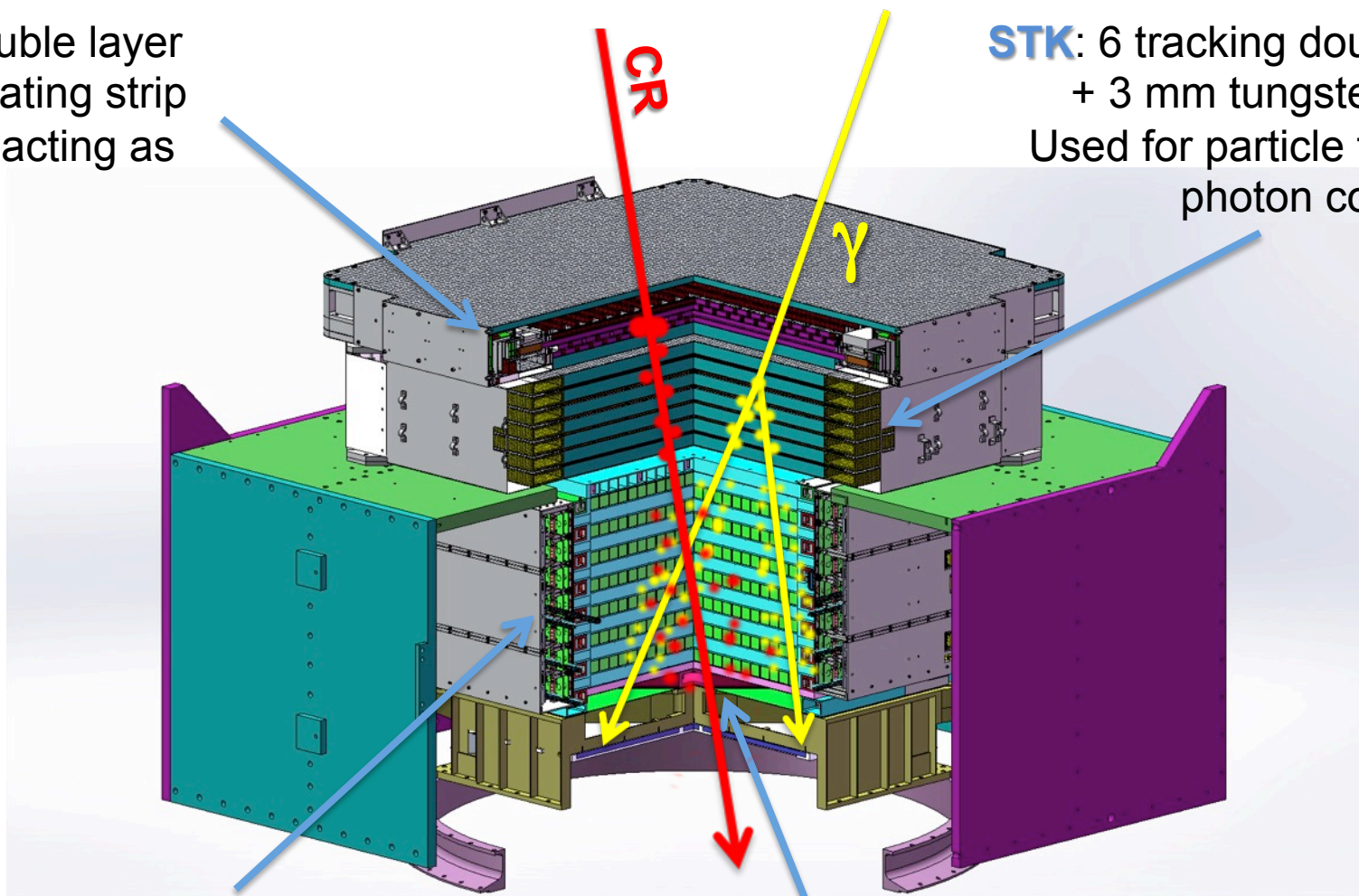
Detection of
2 GeV - 10 TeV e/ γ
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with excellent energy resolution , tracking precision
and particle identification capabilities

- Exotica and “unexpected” , e.g. GW e.m. counterpart in the FoV

The detector

PSD: double layer of scintillating strip detector acting as ACD

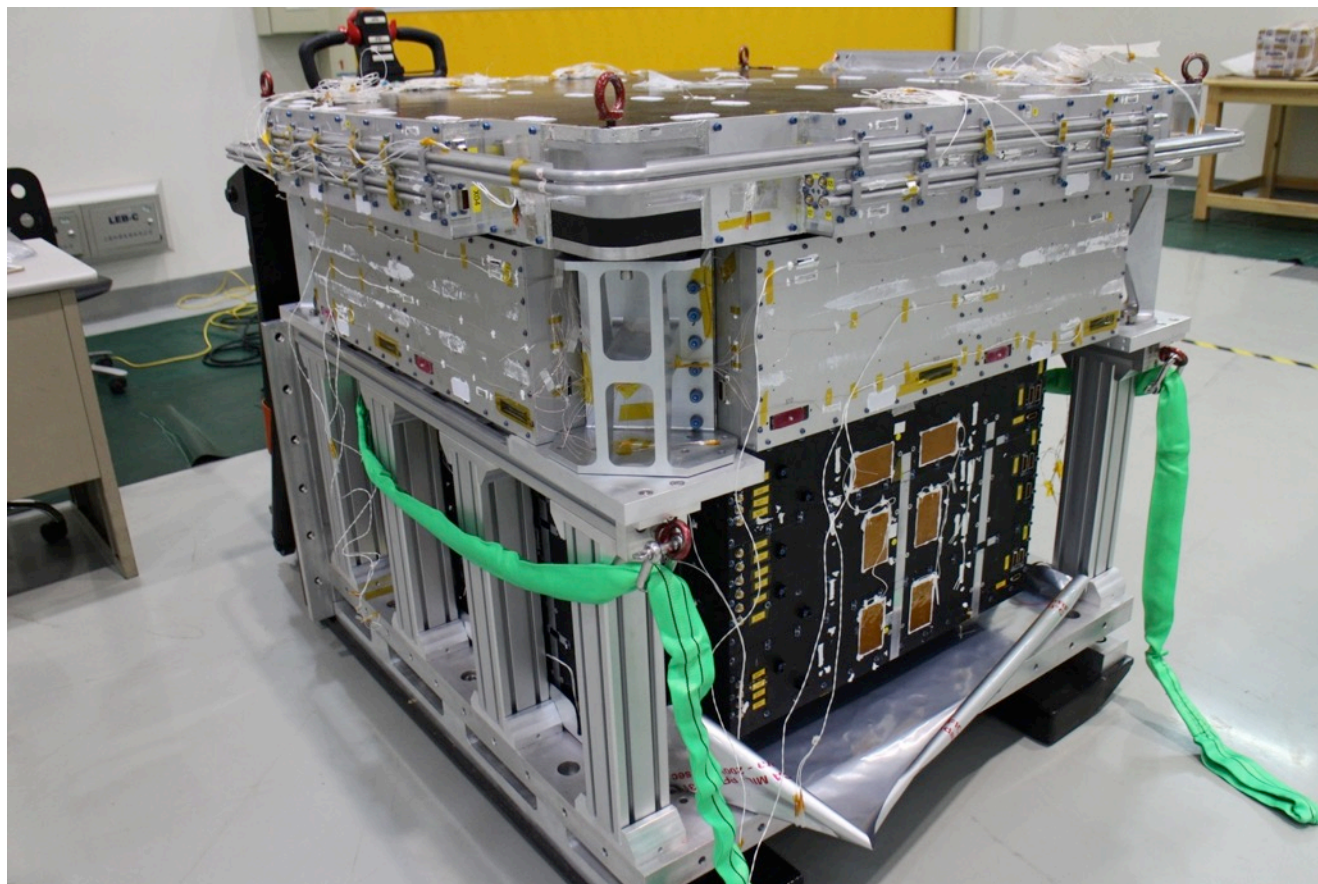
STK: 6 tracking double layer + 3 mm tungsten plates. Used for particle track and photon conversion



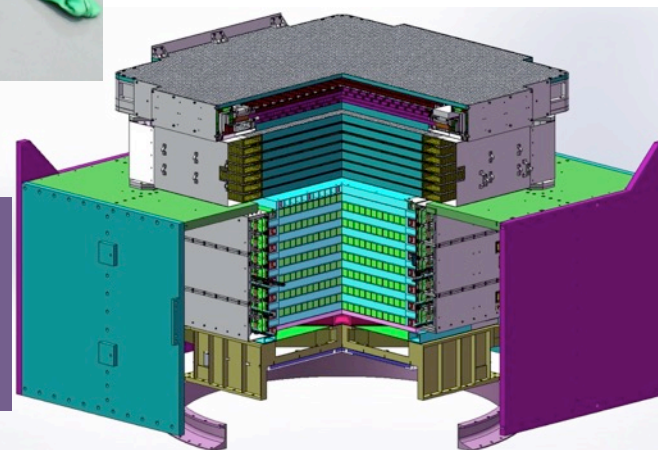
BGO: the calorimeter made of 308 BGO bars in hodoscopic arrangement (~ 31 radiation length). Performs both energy measurements and trigger

NUD: it's complementary to the BGO by measuring the thermal neutron shower activity. Made up of boron-doped plastic scintillator

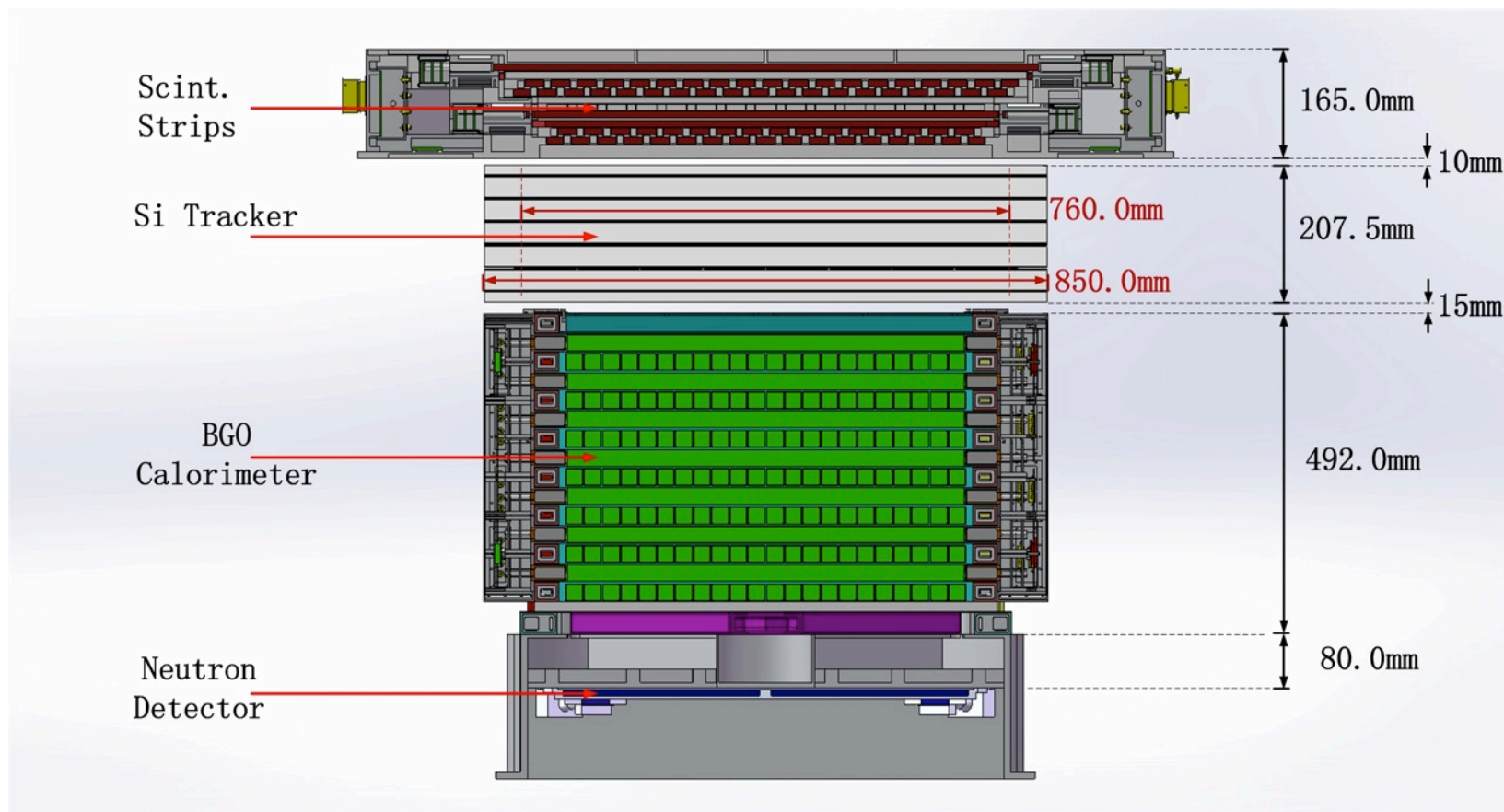
The detector (1)



**W converter ($1.43 X_0$) + thick calorimeter ($31 X_0$)
+ precise tracking + charge measurement \Rightarrow
high energy γ -ray, electron and CR telescope**

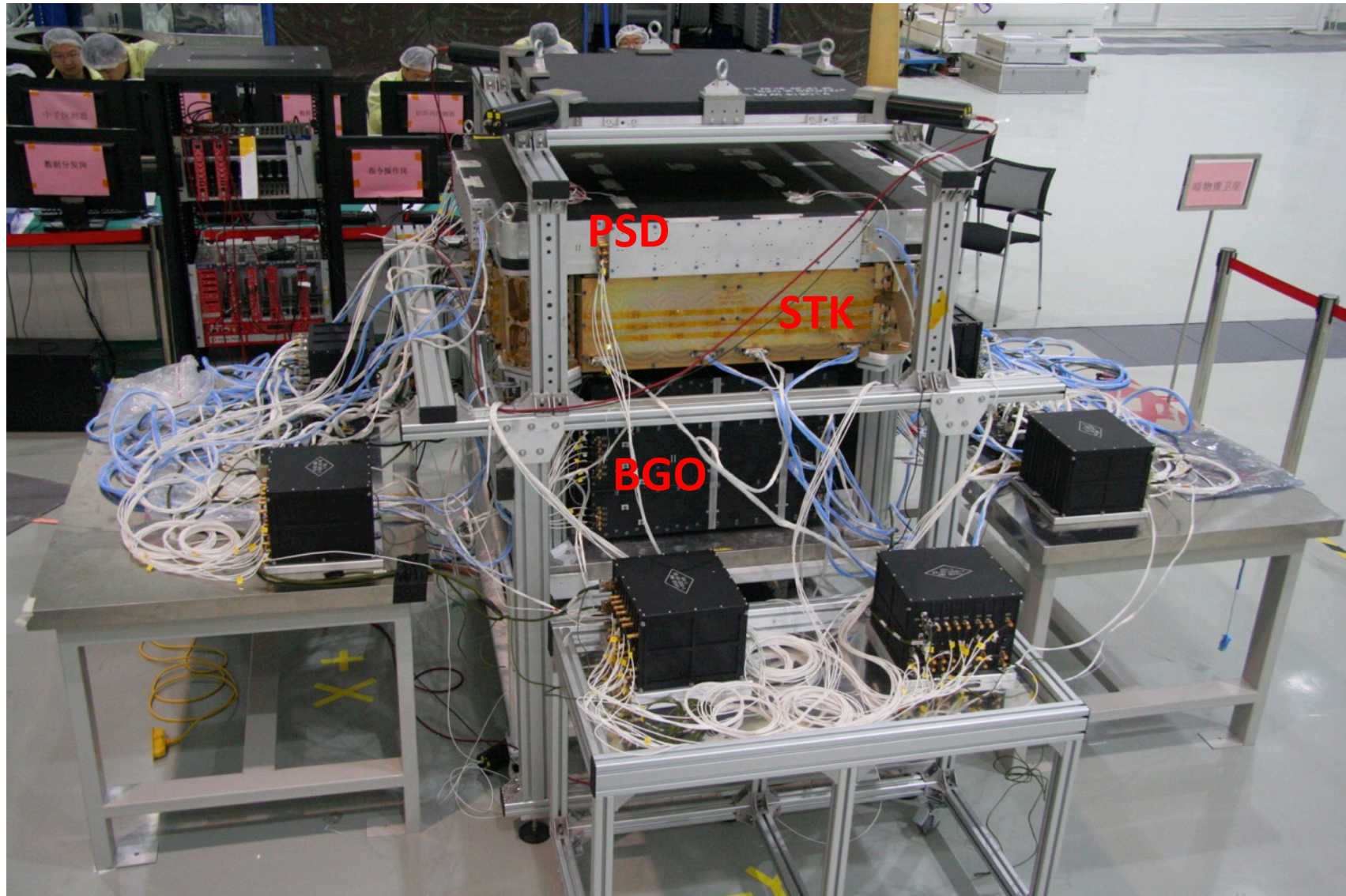


The detector (2)



Mass: 1400 Kg
Power: ~ 400 W
Data: 12 Gbyte/day
Lifetime: 5 years

FM final integration (06/2015)



The collaboration

- **CHINA**

- Purple Mountain Observatory, CAS, Nanjing
- Institute of High Energy Physics, CAS, Beijing
- National Space Science Center, CAS, Beijing
- University of Science and Technology of China, Hefei
- Institute of Modern Physics, CAS, Lanzhou

Prof. Jin Chang



- **ITALY**

- INFN Perugia and University of Perugia
- INFN Bari and University of Bari
- INFN Lecce and University of Salento



- **SWITZERLAND**

- University of Geneva



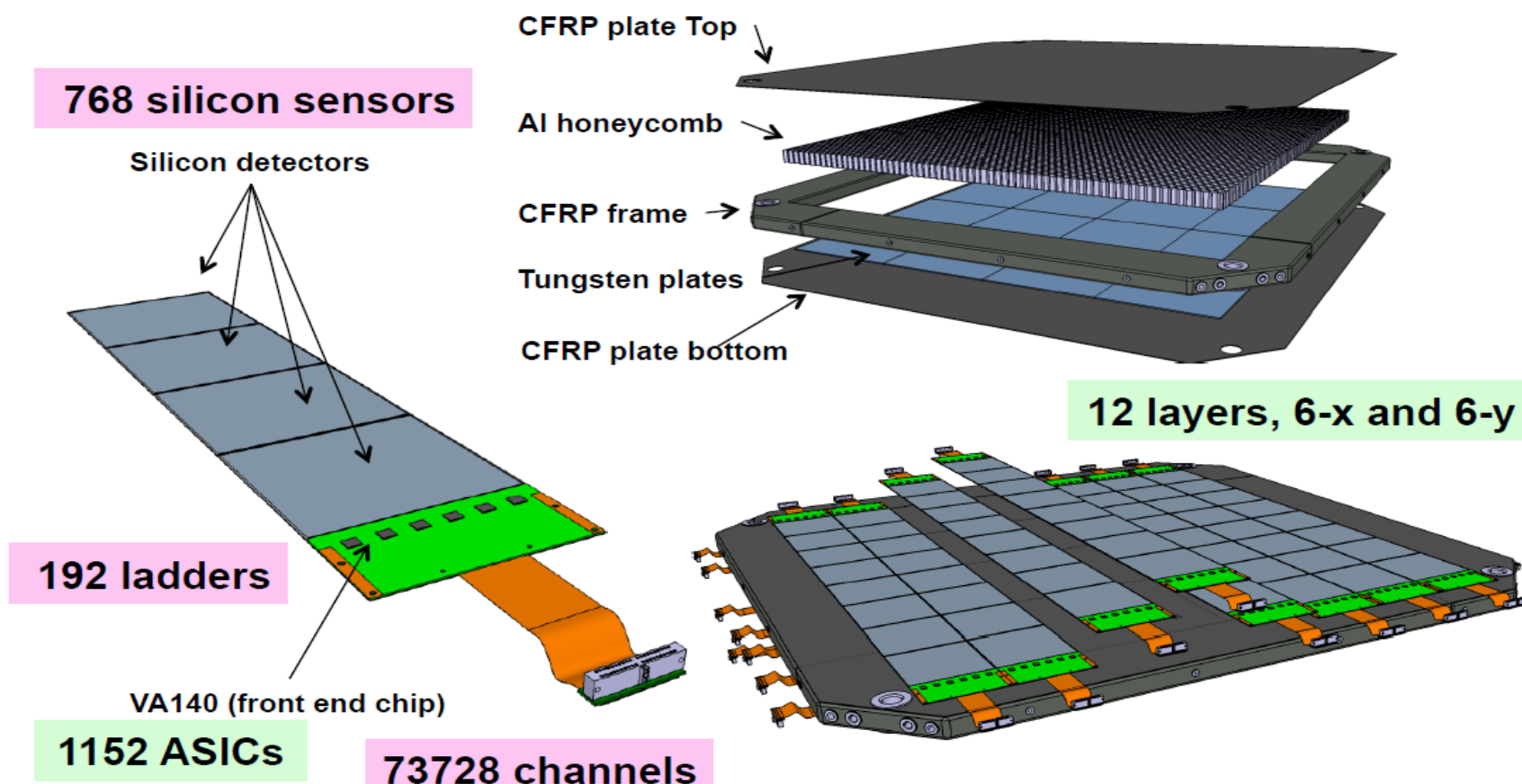
MoU signed on April 2013



CAS (China), Geneva Uni (Switzerland), INFN (Italy)



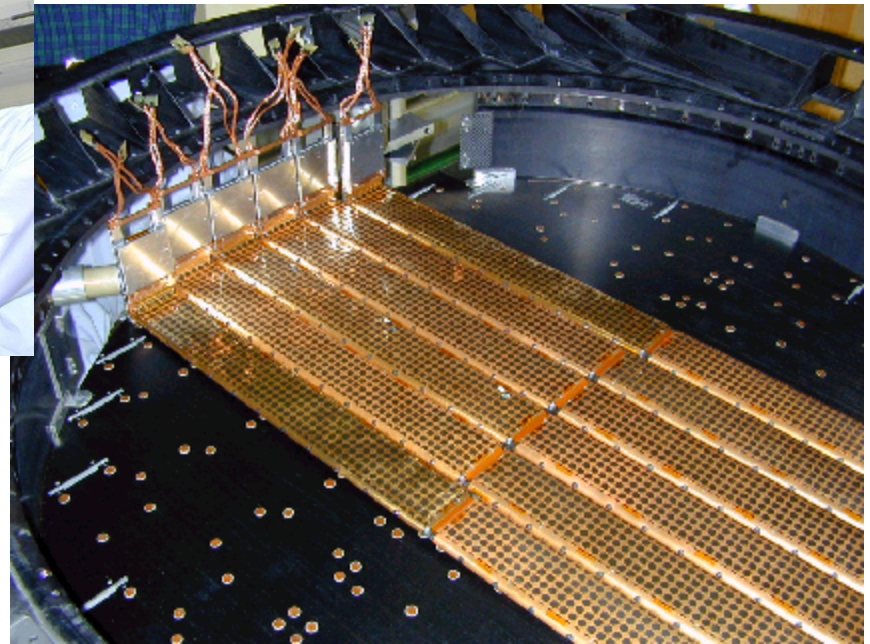
The Silicon Tracker (STK)



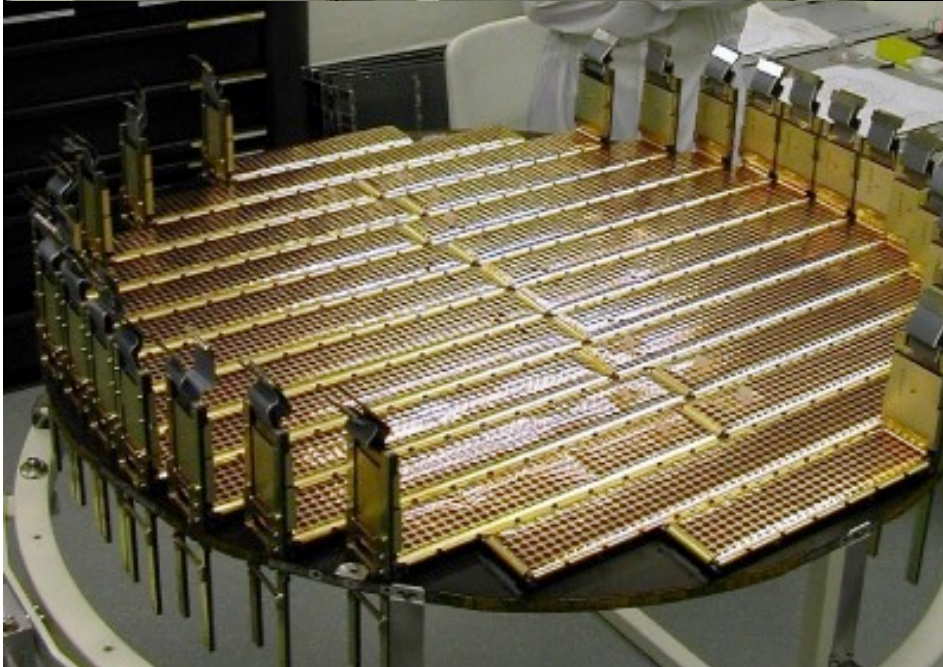
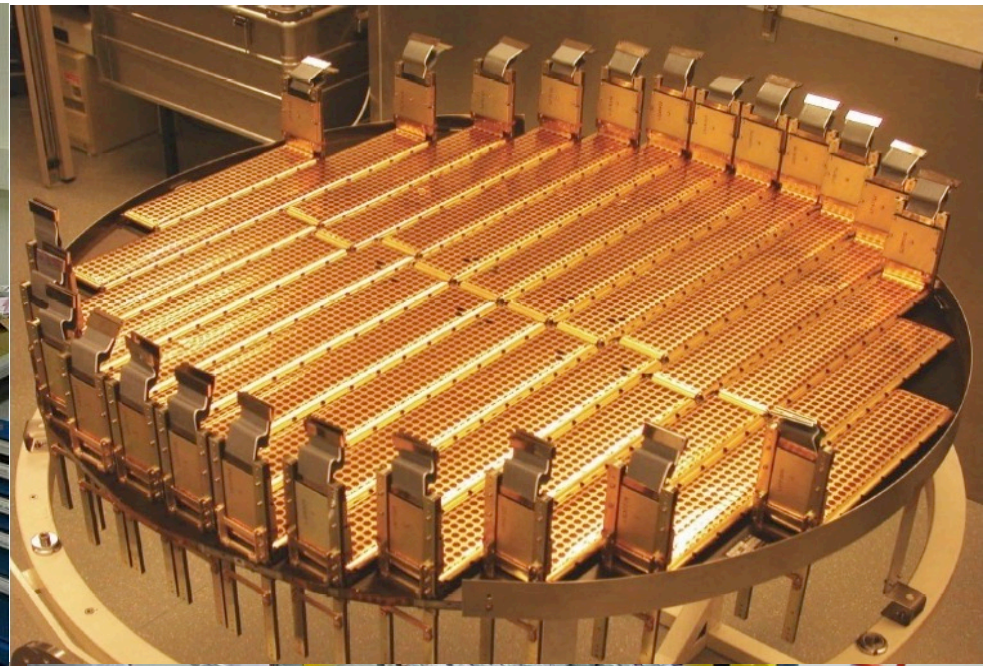
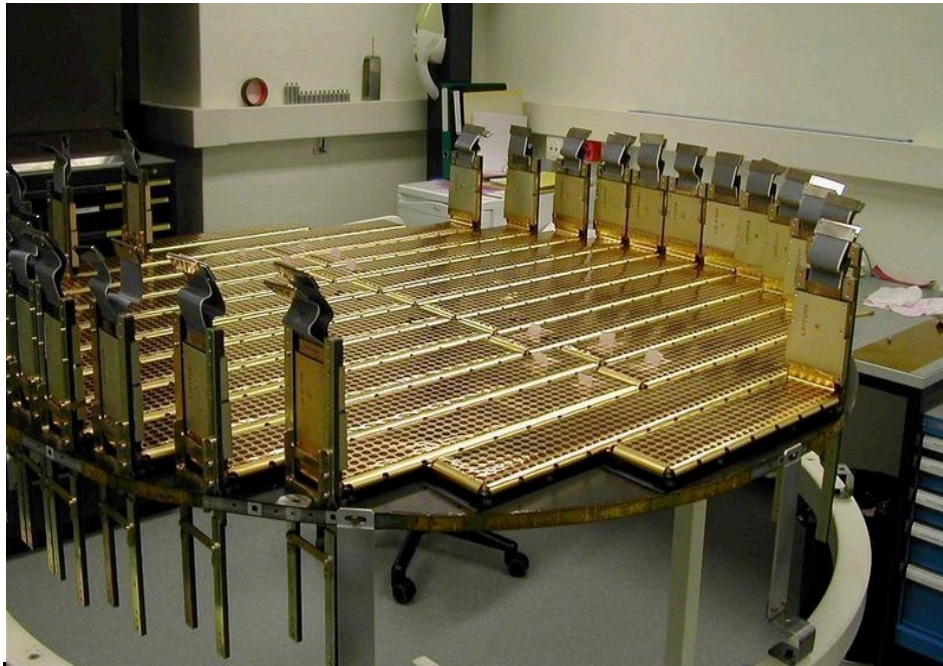
- 48 μm wide Si strips with 121 μm pitch
- (95 × 95 × 0.32 mm³) Silicon Strip Detectors (SSD) with 768 strips
- One ladder composed by 4 Silicon Strip Detectors (SSD)
- 16 Ladders per layer (76 cm × 76 cm)
- 12 layers (6x + 6y)

Analog Readout of each second strip:
384 channels / SSD- Ladder
Charge sharing

AMS-01 Silicon Tracker

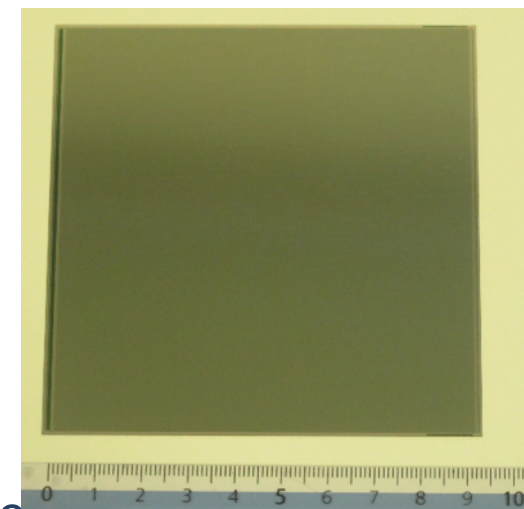


AMS-02: 9 planes with 200,000 channels aligned to 10 microns

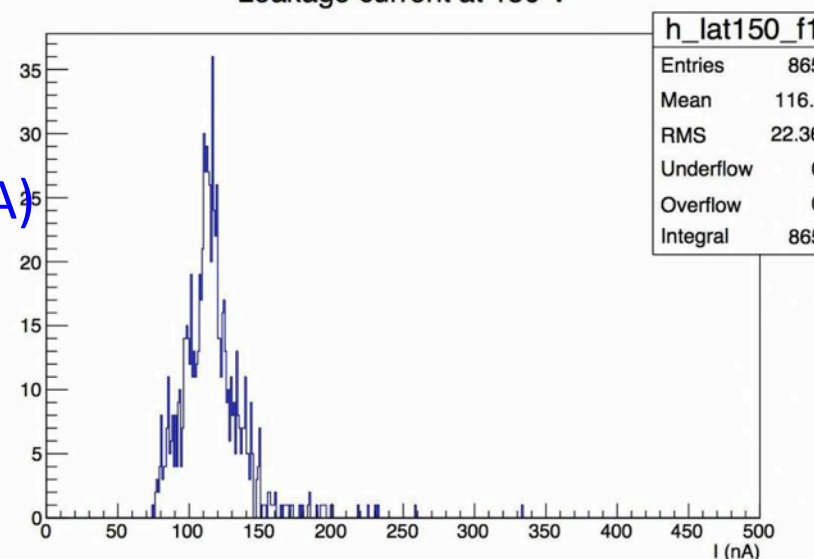


STK Silicon Sensors

- Single-sided Silicon strip detectors produced by Hamamatsu
 - 9.5 x 9.5 cm², 768 strips, 121 μm pitch (AGILE geometry)
 - 320 μm thick (AGILE: 410 μm)
 - Resistivity 5-8 kΩ, V_{fd} 10-80 V
 - Total strip capacitance 2.1 pF/cm
- 150 SSDs for EQM (Engineering and Qualification Model)
- 865 SSDs for FM (Flight Model)
 - Excellent quality
 - $\langle I_{\text{leak}} \rangle \sim 120 \text{ nA @ 150V}$ (spec: $< 900 \text{ nA}$)
 - V_{fd} < 50 V
 - Very few bad channels
 - Cut precision: $\sim \text{few } \mu\text{m}$

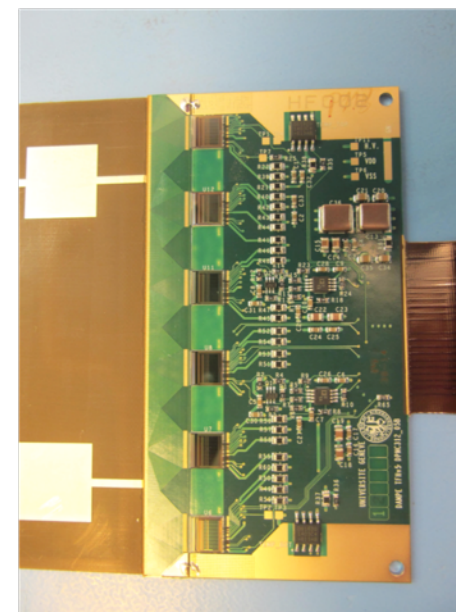
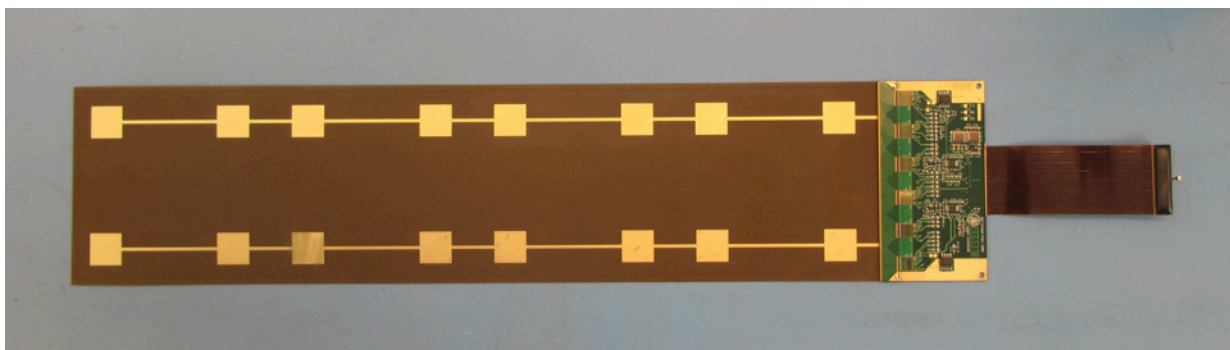


Leakage current at 150 V



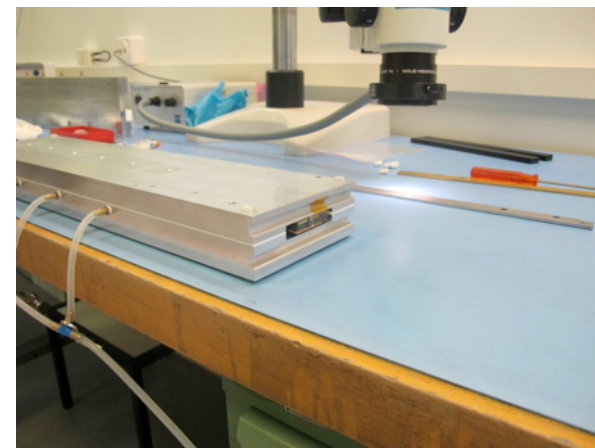
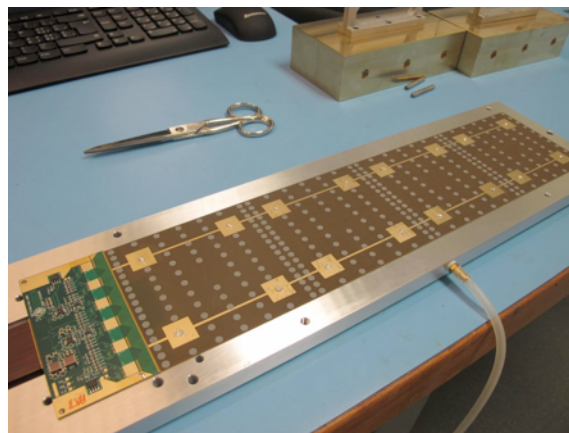
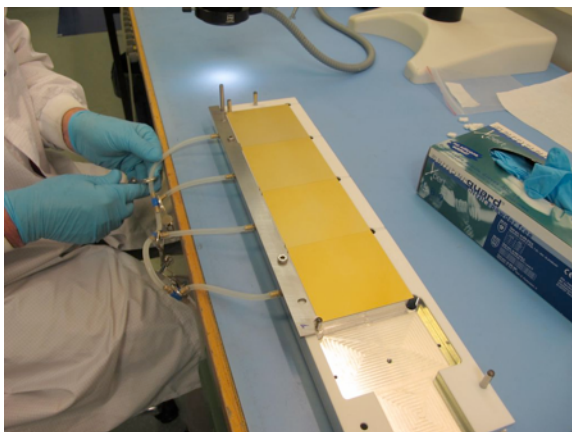
STK Readout Electronics

- Readout every other strip, readout pitch 242 μm
- ASIC: VA140 from IDEAS, updated version of VA64hdr of AMS-02
 - Low power (0.3 mW/channel) and large dynamic range (200 fC)
 - Analog readout
 - Charge measurement
 - Better position resolution with charge sharing
- Tracker Front-end Hybrid (TFH)
 - Thin bias circuit integrated with a PCB housing 6 ASICs, and a readout cable (“pigtail”)
 - Support structure for the SSDs
 - Vias and copper bands for heat transfer

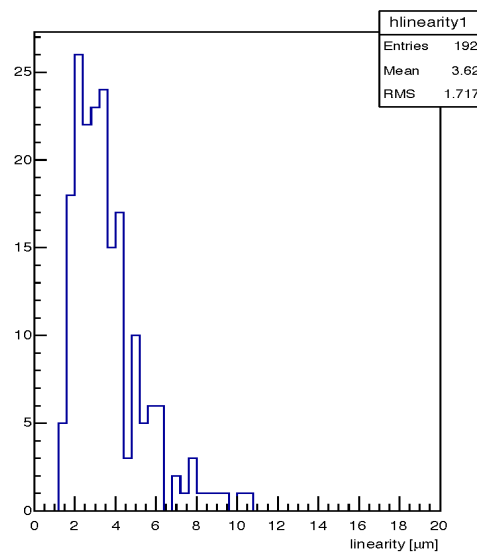


STK Silicon ladders

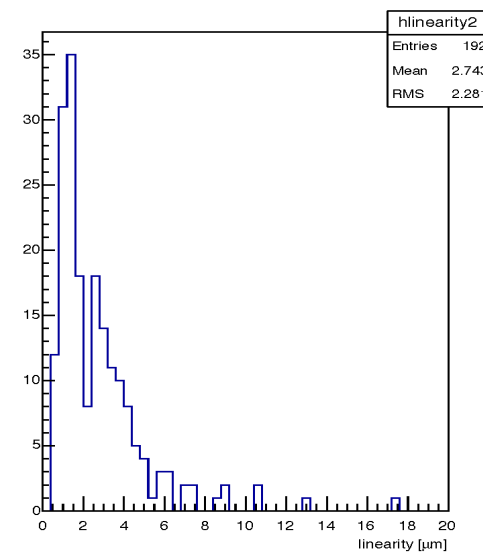
- Precise jigs to assemble (align, glue and bond) 4 sensors to form a ladder
 - require 20 μm alignment precision \Rightarrow achieved, most ($\sim 97\%$) $< 10 \mu\text{m}$



Linearity top side

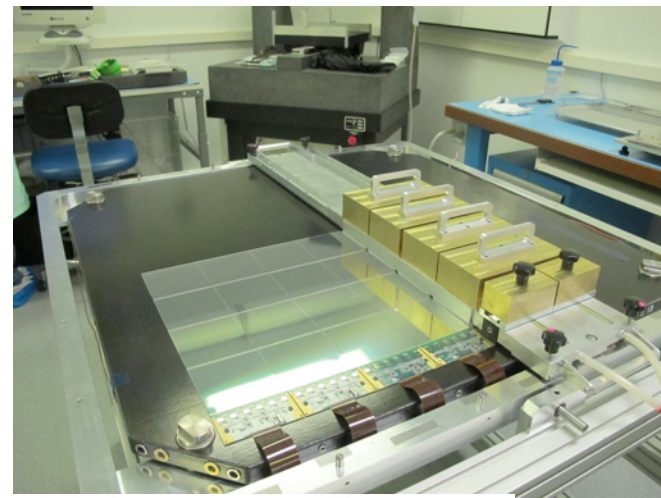
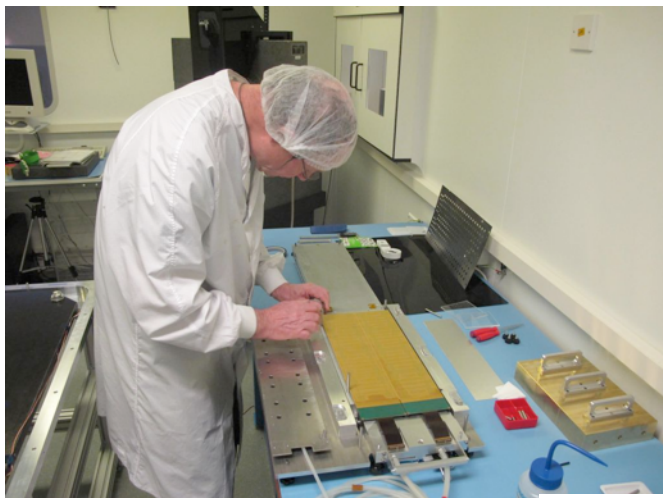


Linearity bottom side

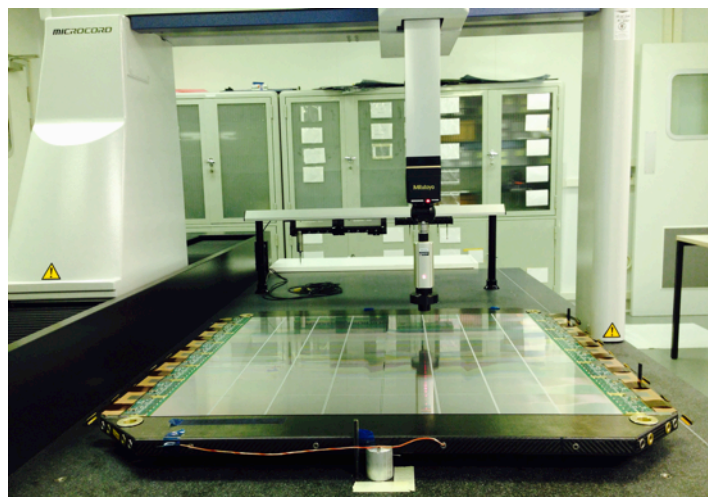


STK Tracker planes

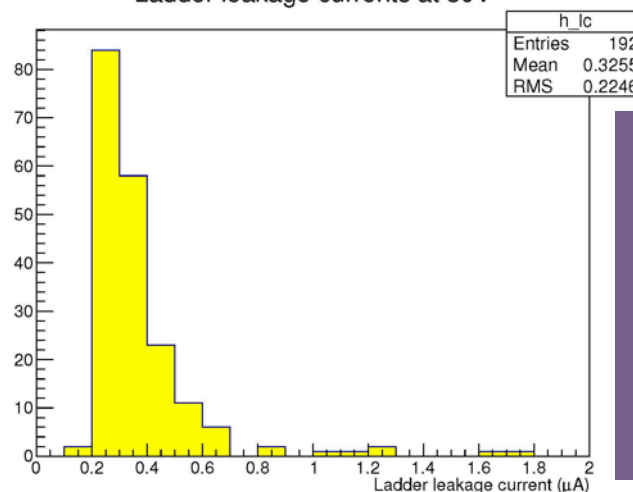
- 16 ladders glued to each surface of the support trays (except top and bottom)



Ladder leakage currents at 80V



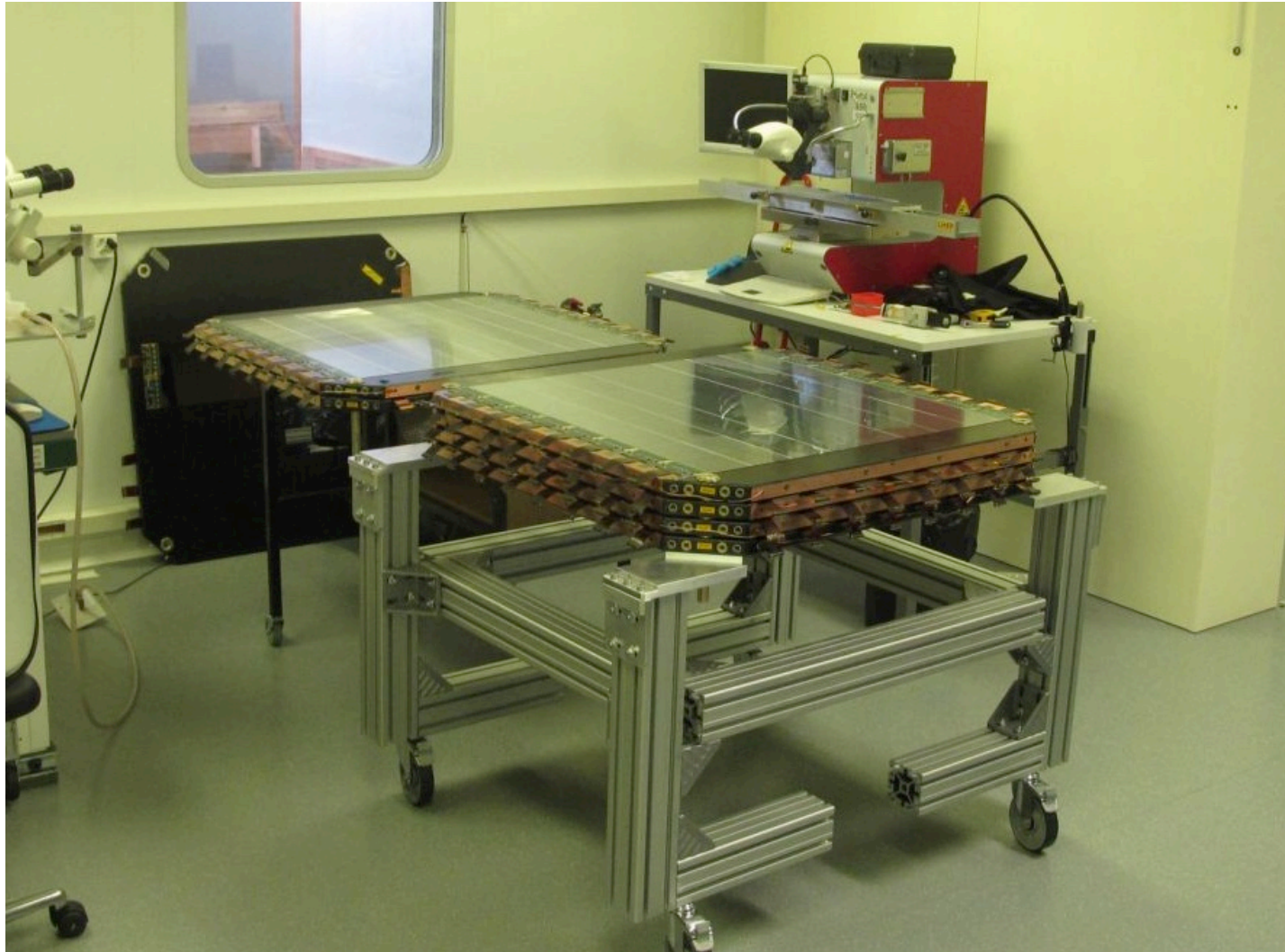
Assembled planes are measured with metrology machine, flatness $\sim 100 \mu\text{m}$



Leakage current of 192 ladders after plane assembly at 80 V

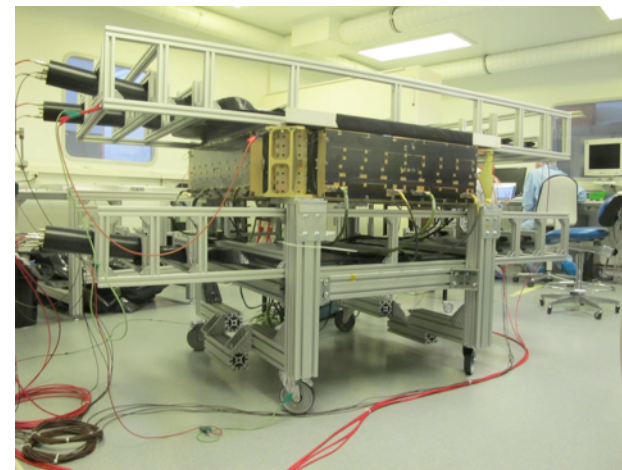
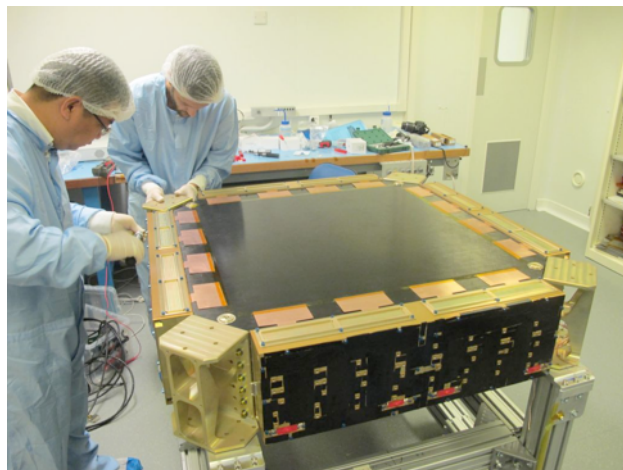
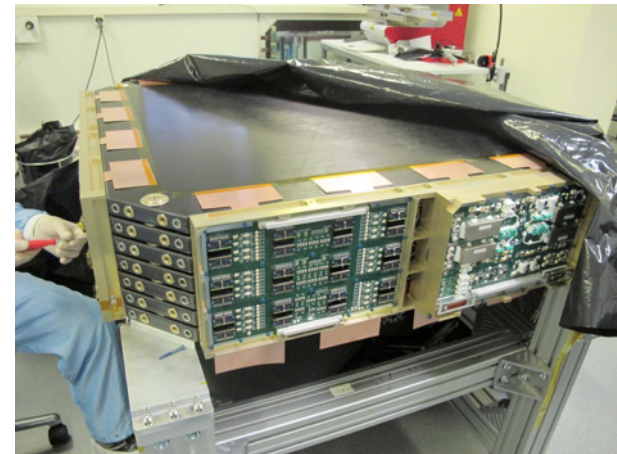
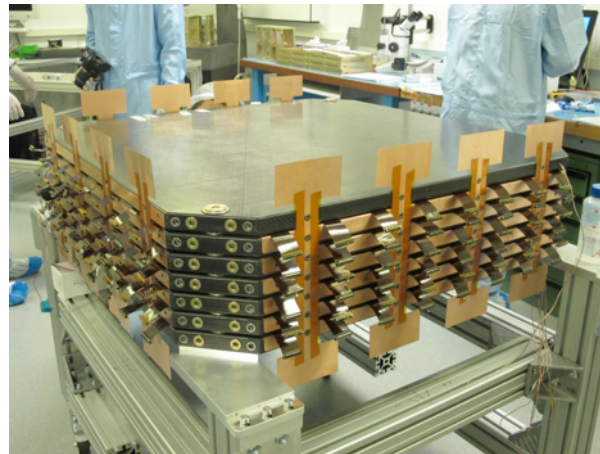
The excellent quality of the silicon sensors maintained through the ladder production and plane assembly processes!

Several layers

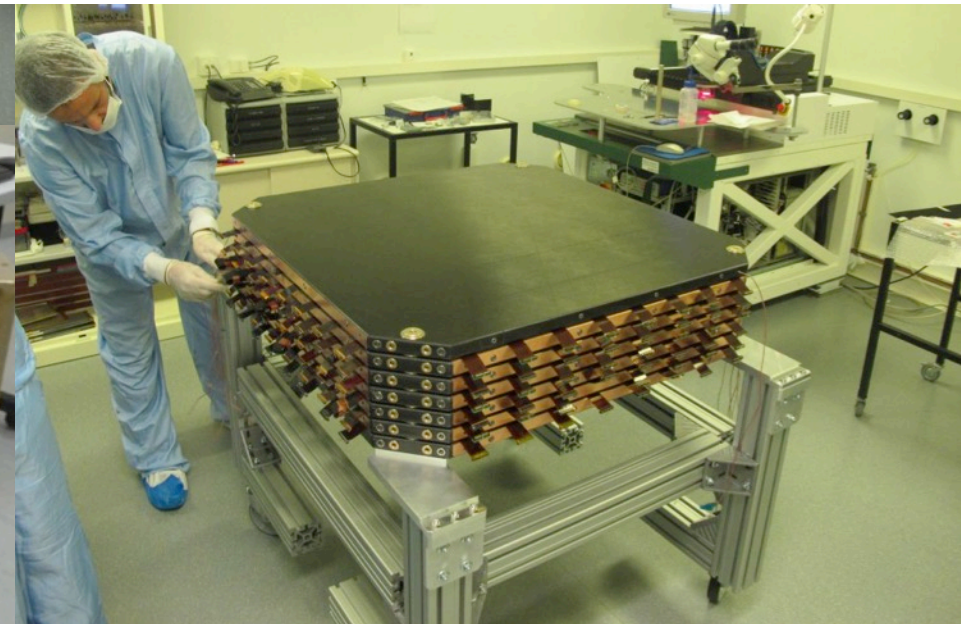
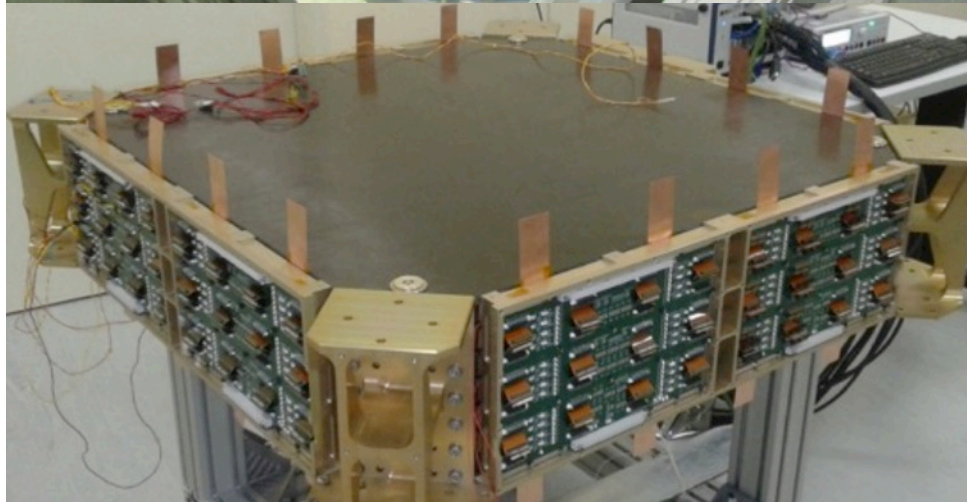
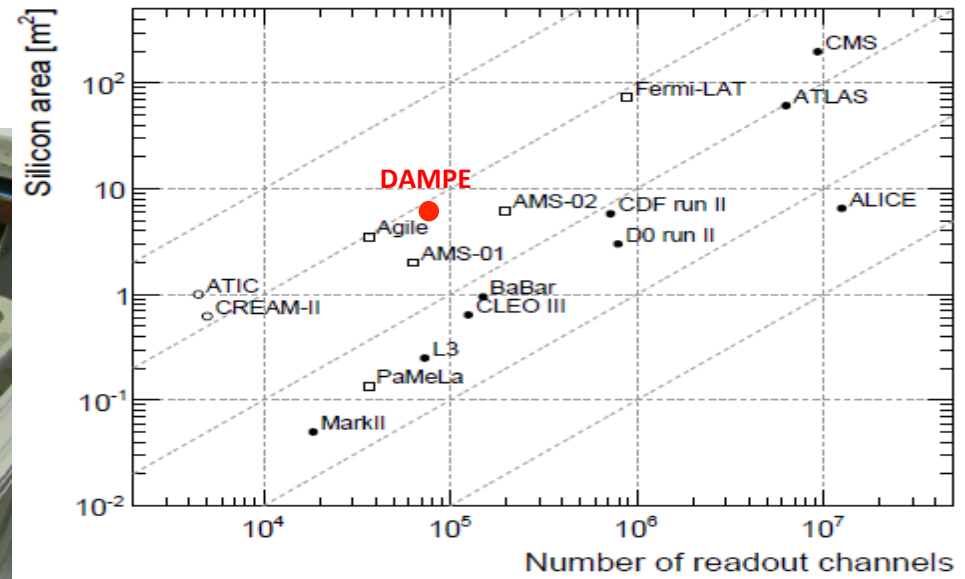
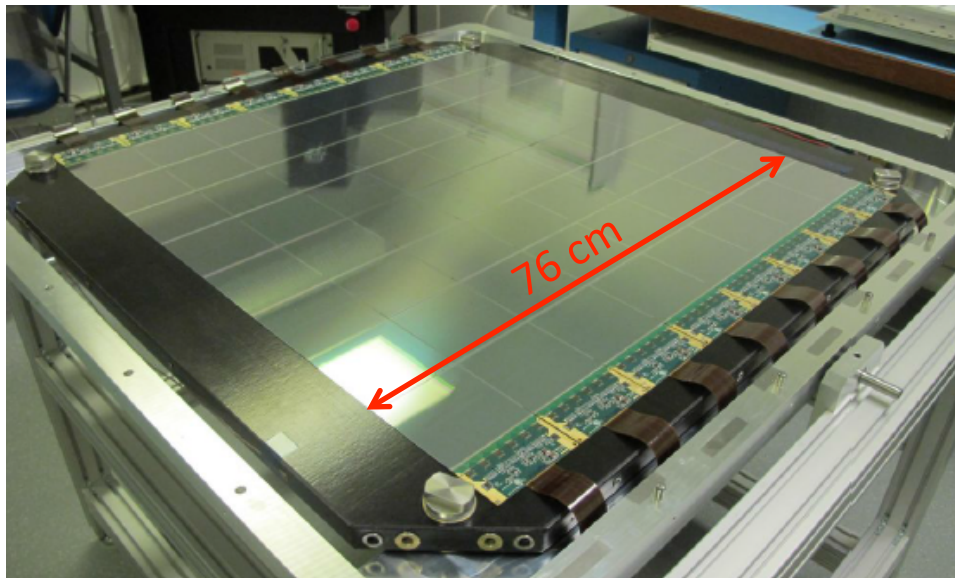


Flight Model Assembly

- The Flight Model assembly has been completed in April 2015
 - Tested with cosmic rays before delivered to China

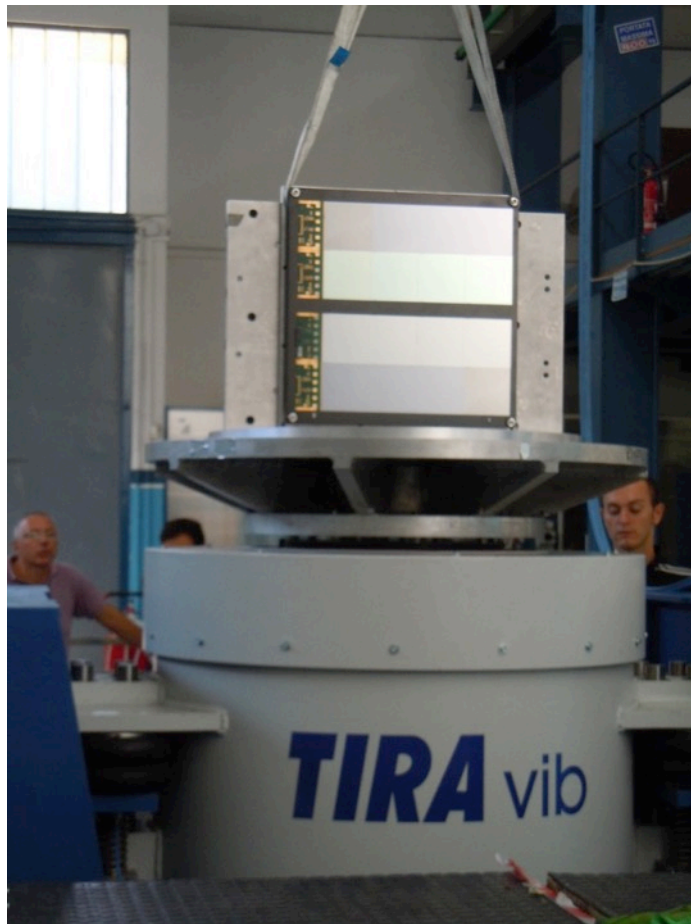


The Silicon Tracker



Engineering and Qualification model

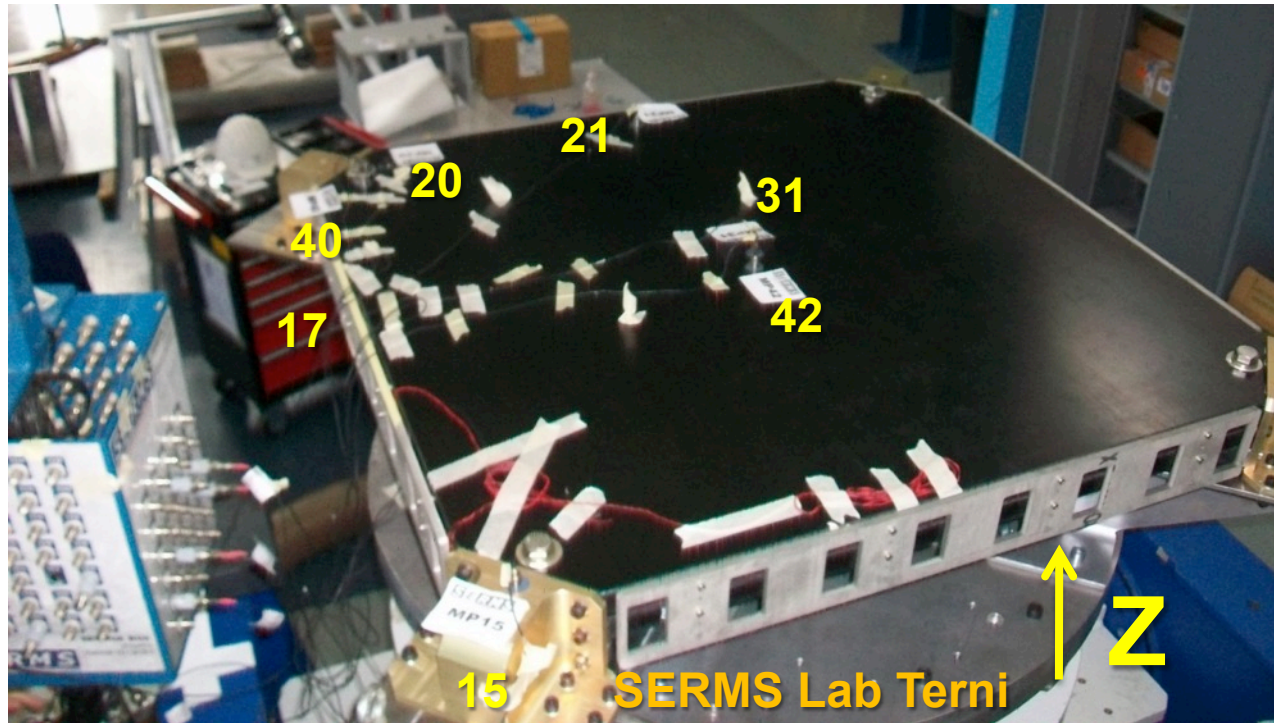
- An EQM has been constructed in July 2014
 - full size model as the final Flight Model (FM), but only 26 ladders with real silicon sensors, the rest with dummy sensors



EQM passed a series of space environmental qualification tests: vibration, acceleration, shock, thermal cycling, thermal vacuum

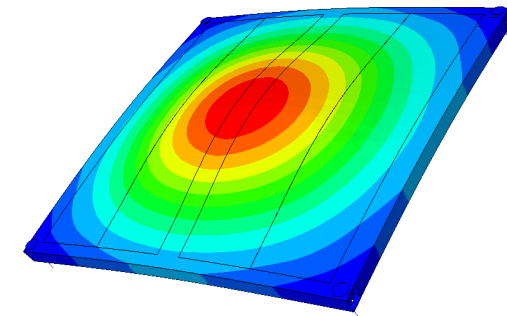
SERMS laboratory
Terni - Italy

DAMPE Mechanical testing:

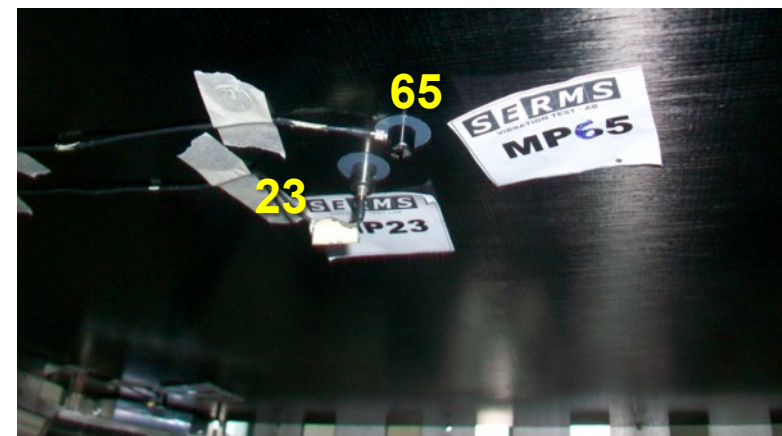


PHASE 5 Hardware:

- 2 bare trays ($d=3,5\text{mm}$)
- 0 ladders
- 0 dummy ladders

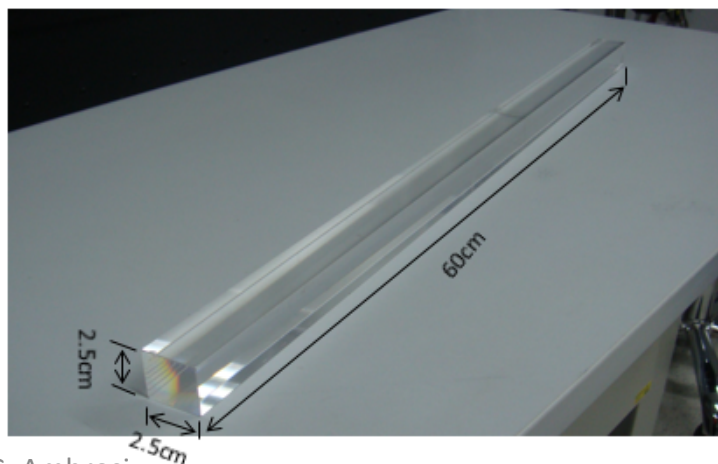
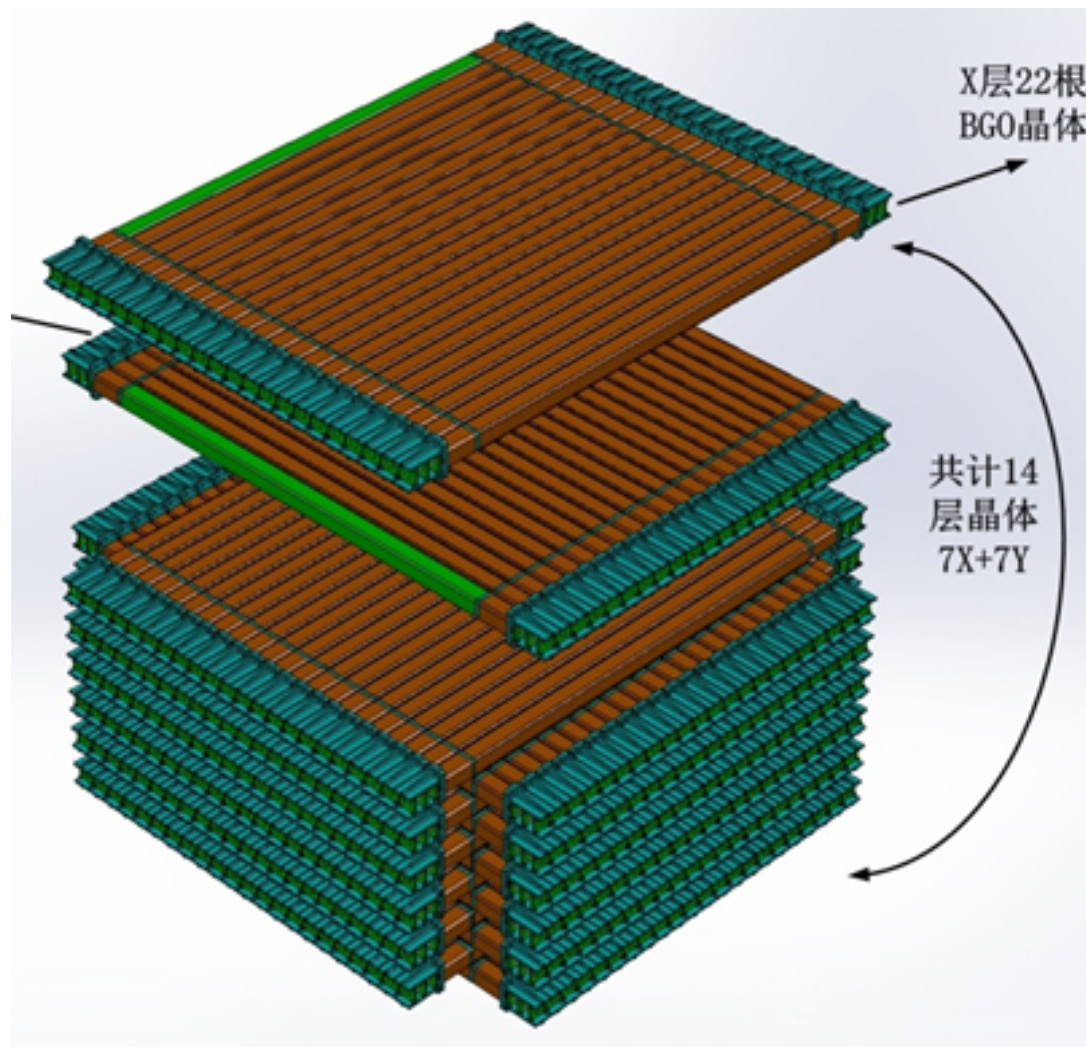


Control Channels: 15 & 17



The CALOrimeter

- 14 layers of 22 BGO bars
 - Dimension of BGO bar: $2.5 \times 2.5 \times 60 \text{ cm}^3$
 - 14 hodoscopic stacking alternating orthogonal layers
 - depth $\sim 32X_0$
- Two PMTs coupled with each BGO crystal bar in two ends



G. Ambrosi

308 bars
616 PMTs



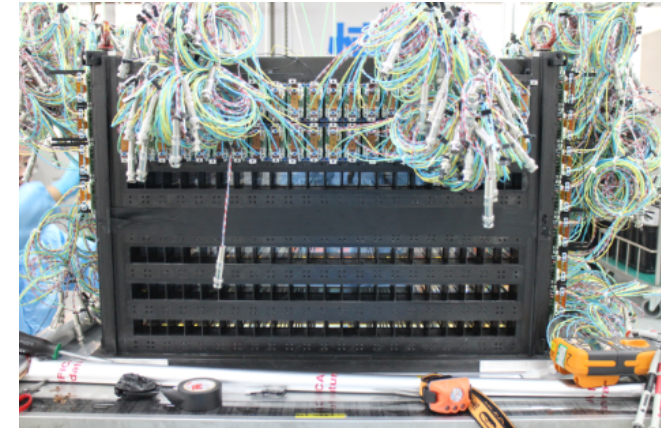
The CALOrimeter -2



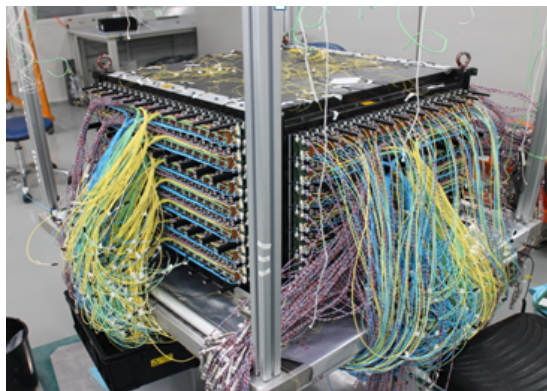
Carbon Fiber Structure



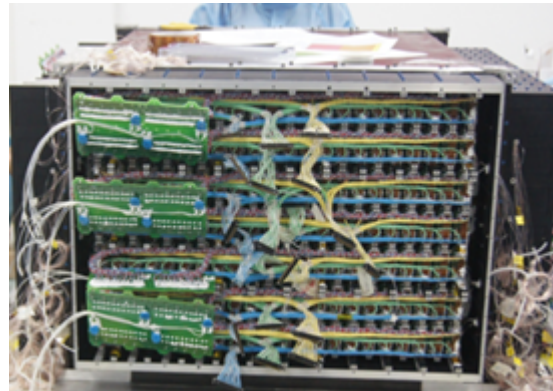
BGO crystal installation



PMT installation



Cable arranging



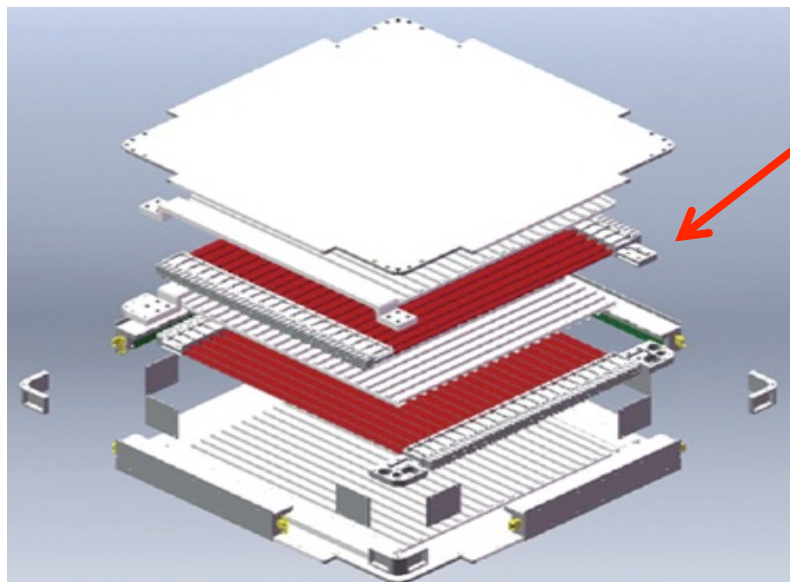
Cable connector



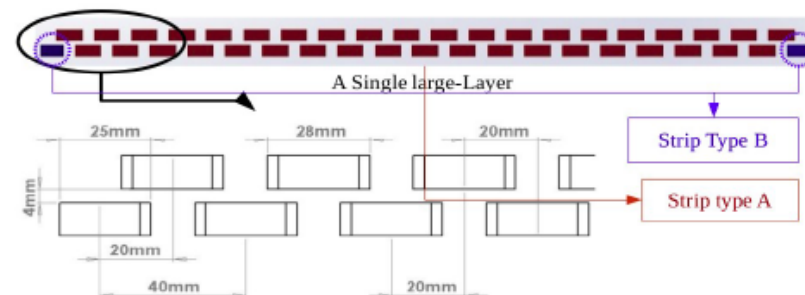
BGO Cal

The PSD and the NUD

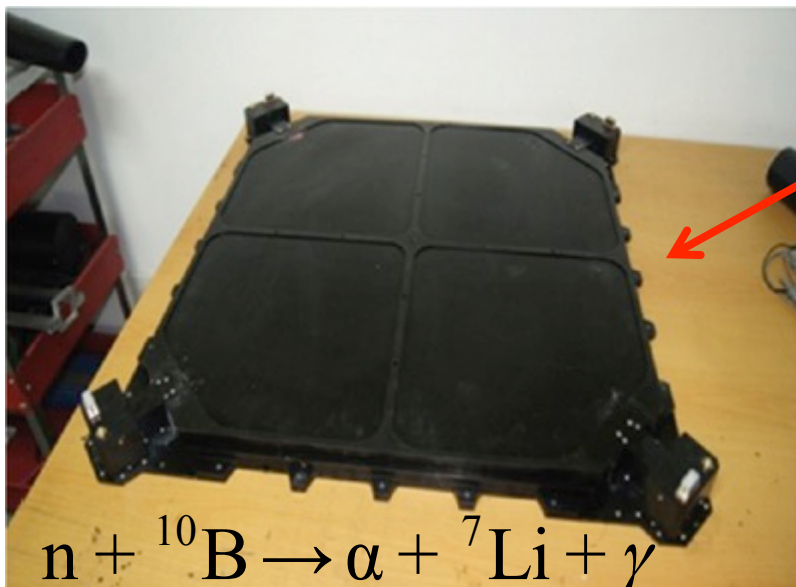
PSD



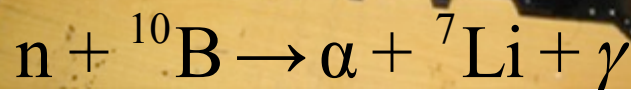
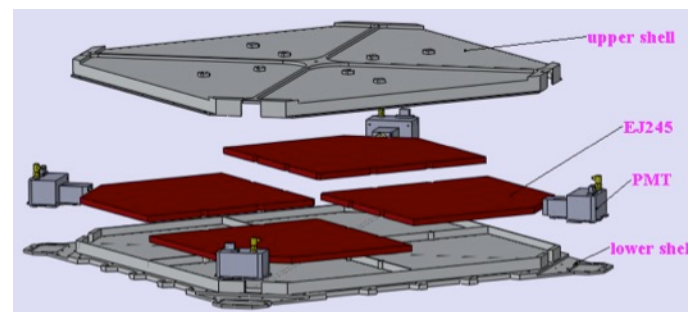
- 1.0 cm thick ,2.8cm wide and 82.0 cm long scintillator strips
- staggered by 0.8 cm in a layer
- 82 cm × 82 cm layers
- 2 layers (x and y)



NUD

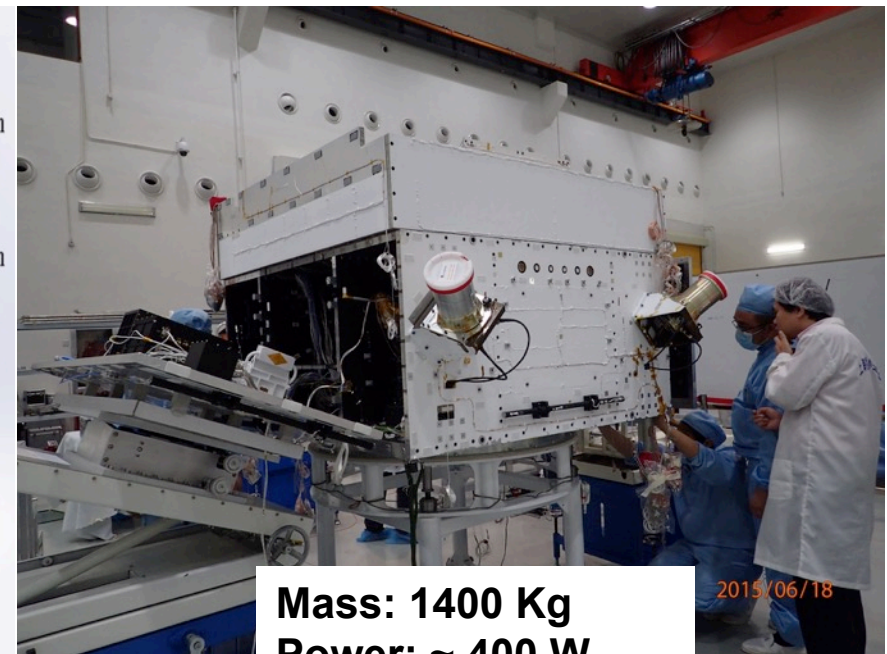
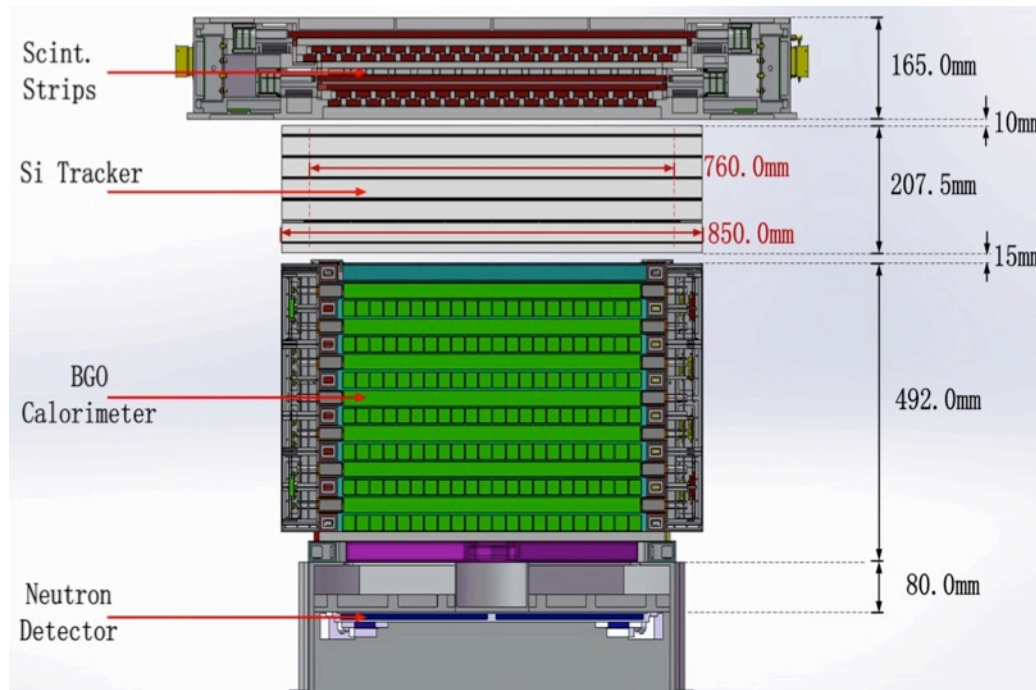


- 4 large area boron-doped plastic scintillators (30 cm × 30 cm × 1 cm)



Comparison with AMS-02 and FERMI

	DAMPE	AMS-02	Fermi LAT
e/ γ Energy res.@100 GeV (%)	1.5	3	10
e/ γ Angular res.@100 GeV (°)	0.1	0.3	0.1
e/p discrimination	10^5	$10^5 - 10^6$	10^3
Calorimeter thickness (X_0)	32	17	8.6
Geometrical accep. (m^2sr)	0.29	0.09	1

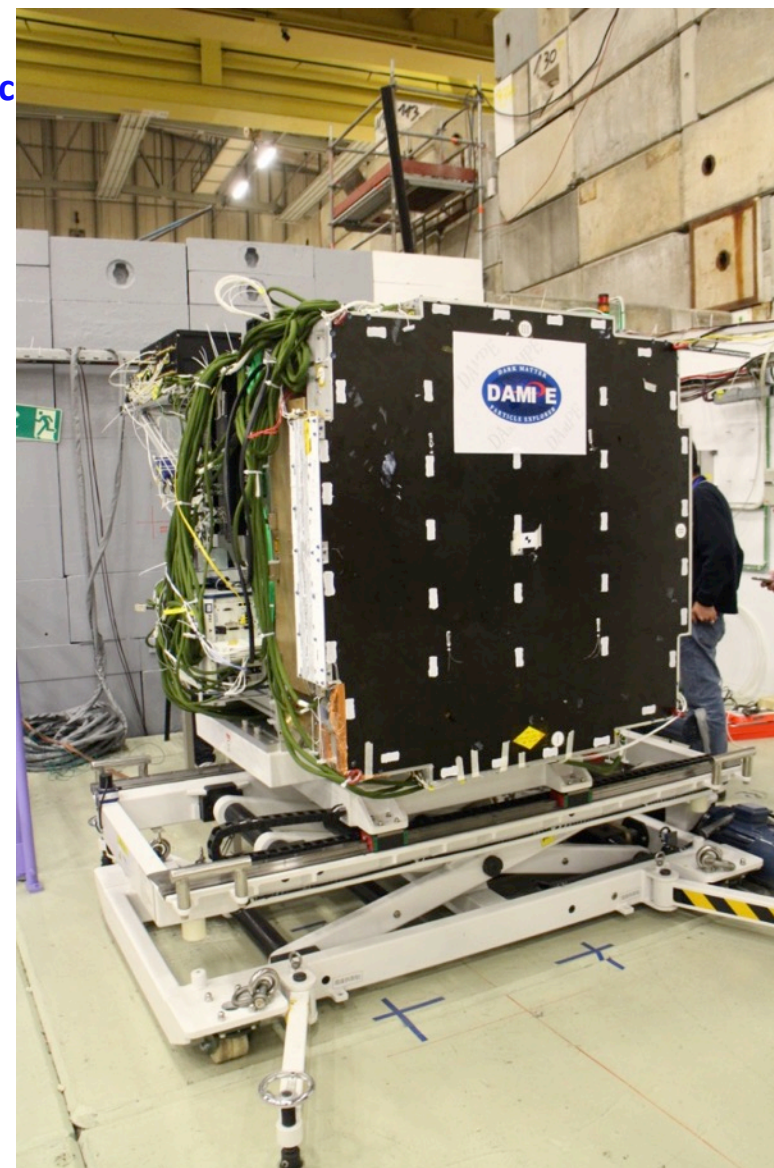


Mass: 1400 Kg
Power: ~ 400 W
Lifetime: > 3 years

2015/06/18

Test beam activity at CERN

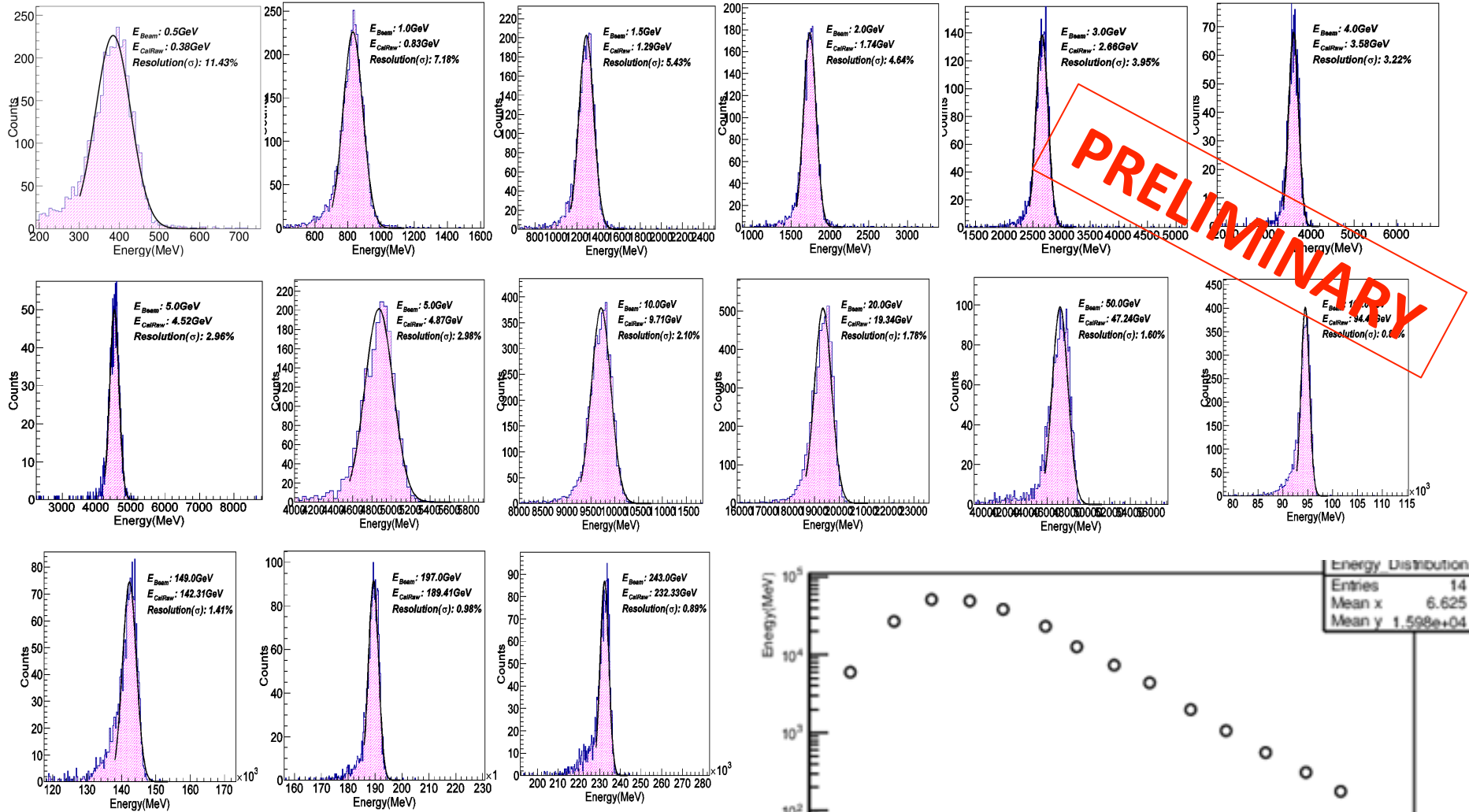
- **14days@PS, 29/10-11/11 2014**
 - e @ 0.5GeV/c, 1GeV/c, 2GeV/c, 3GeV/c, 4GeV/c, 5GeV/c
 - p @ 3.5GeV/c, 4GeV/c, 5GeV/c, 6GeV/c, 8GeV/c, 10GeV/c
 - π^- @ 3GeV/c, 10GeV/c
 - γ @ 0.5-3GeV/c
- **8days@SPS, 12/11-19/11 2014**
 - e @ 5GeV/c, 10GeV/c, 20GeV/c, 50GeV/c, 100GeV/c, 150GeV/c, 200GeV/c, 250GeV/c
 - p @ 400GeV/c (SPS primary beam)
 - γ @ 3-20GeV/c
 - μ @ 150GeV/c,
- **17days@SPS, 16/3-1/4 2015**
 - Fragments: 66.67-88.89-166.67GeV/c
 - Argon: 30A- 40A- 75AGeV/c
 - Proton: 30GeV/c, 40GeV/c
- **21days@SPS, 10/6-1/7 2015**
 - Primary Proton: 400GeV/c
 - Electrons @ 20, 100, 150 GeV/c
 - γ @ 50, 75 , 150 GeV/c
 - μ @ 150 GeV /c
 - π^+ @10, 20, 50, 100 GeV/c
- **10days@SPS, 11/11-20/11 2015**
 - Pb 30AGeV/c (and fragments) (HERD)
- **6days@SPS, 20/11-25/11 2015**
 - Pb 030 AGeV/c (and fragments)



DAMPE first flight: EQM beam test



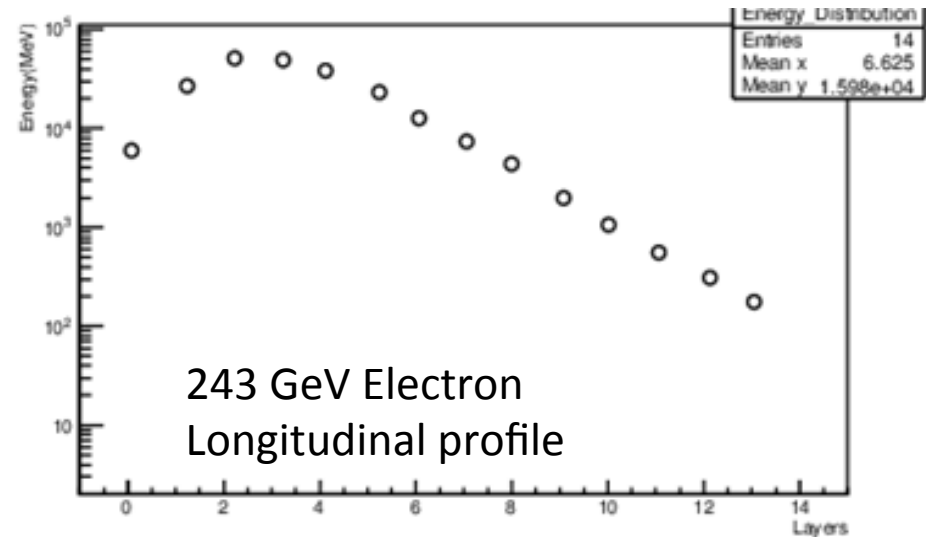
Test beam activity at CERN: electrons



Electrons

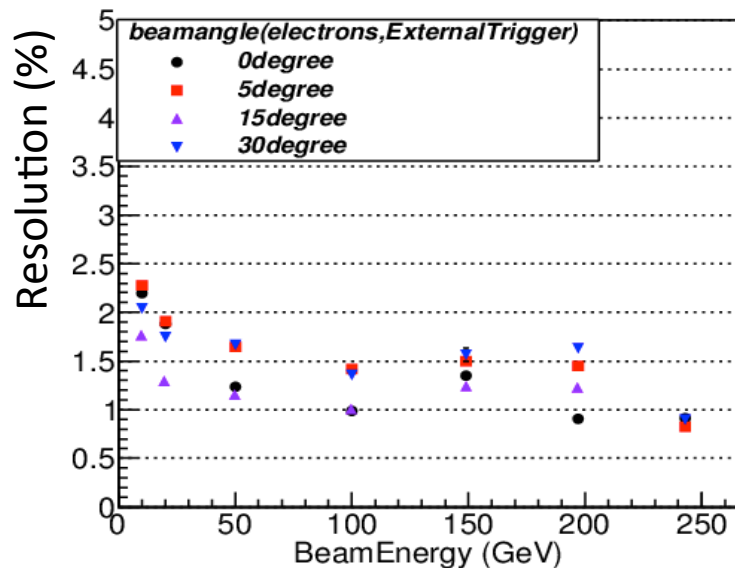
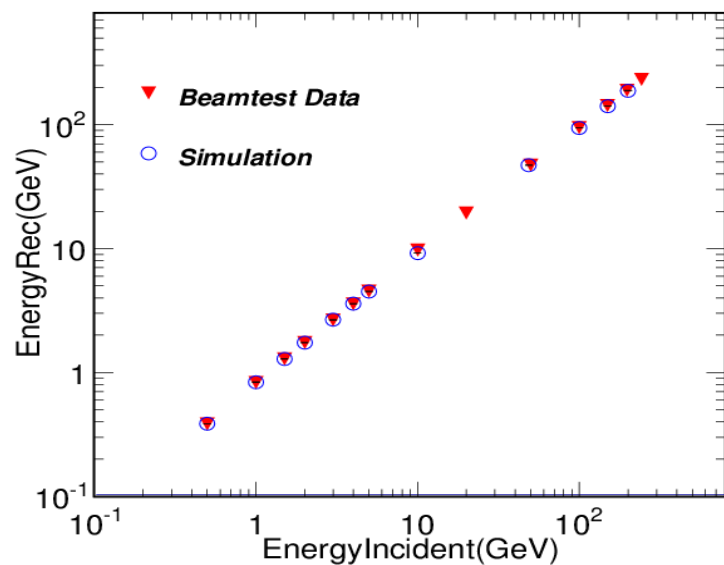
0.5, 1, 1.5, 2, 3, 4, 5 GeV @ PS

5, 10, 20, 50, 100, 149, 197, 243 GeV @ SPS

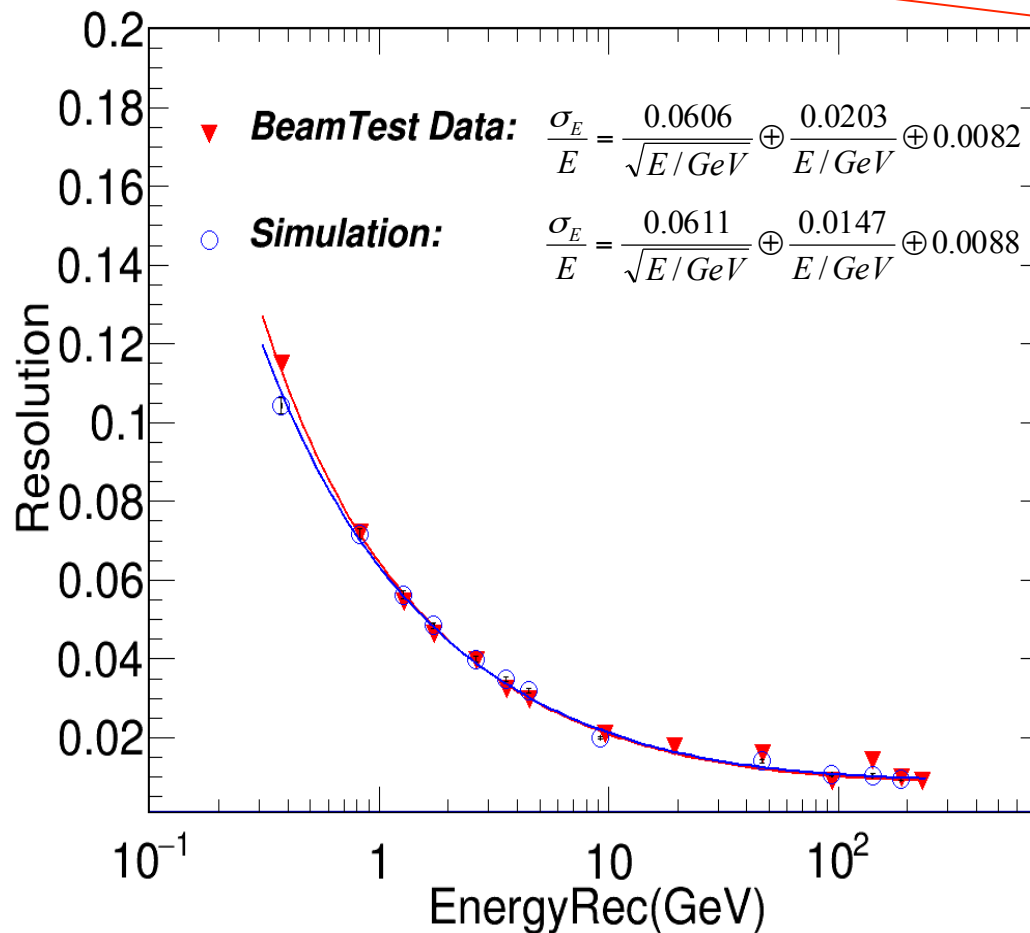


243 GeV Electron
Longitudinal profile

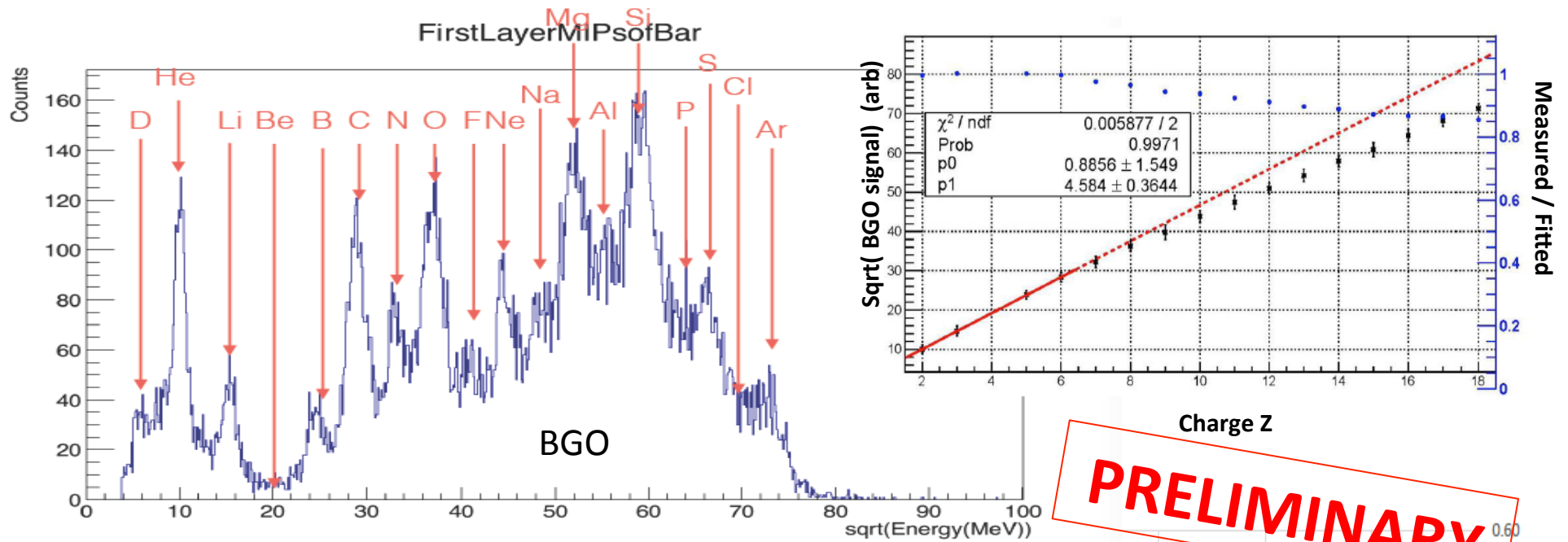
Test beam activity at CERN: electrons



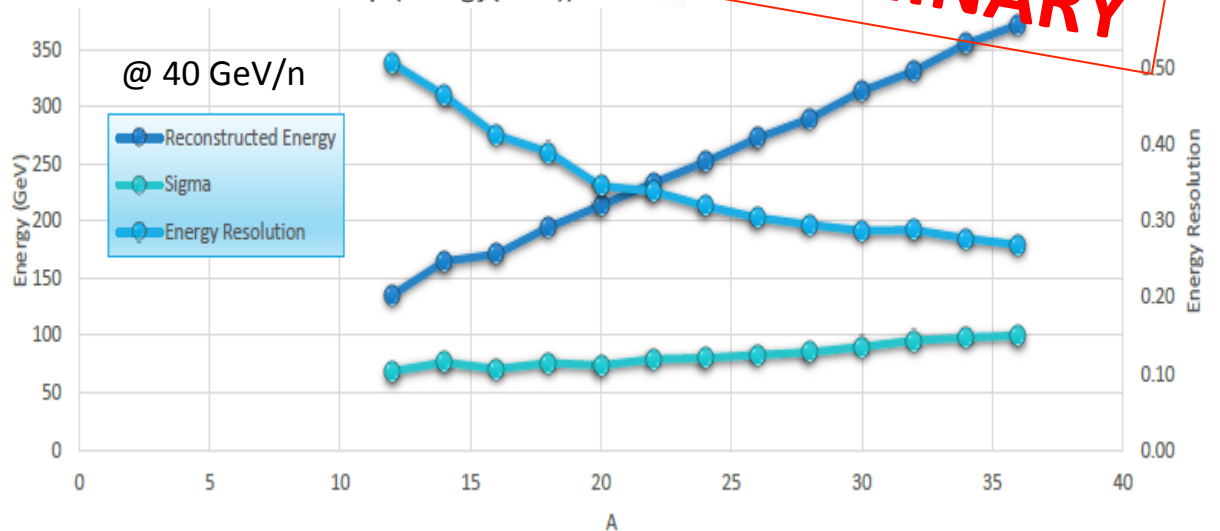
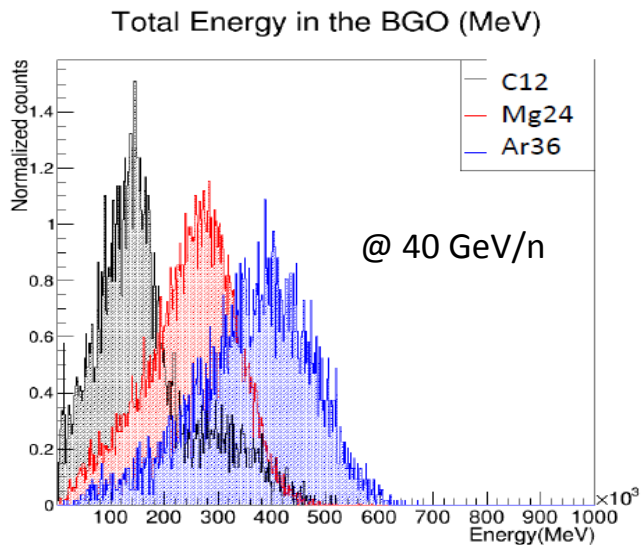
PRELIMINARY



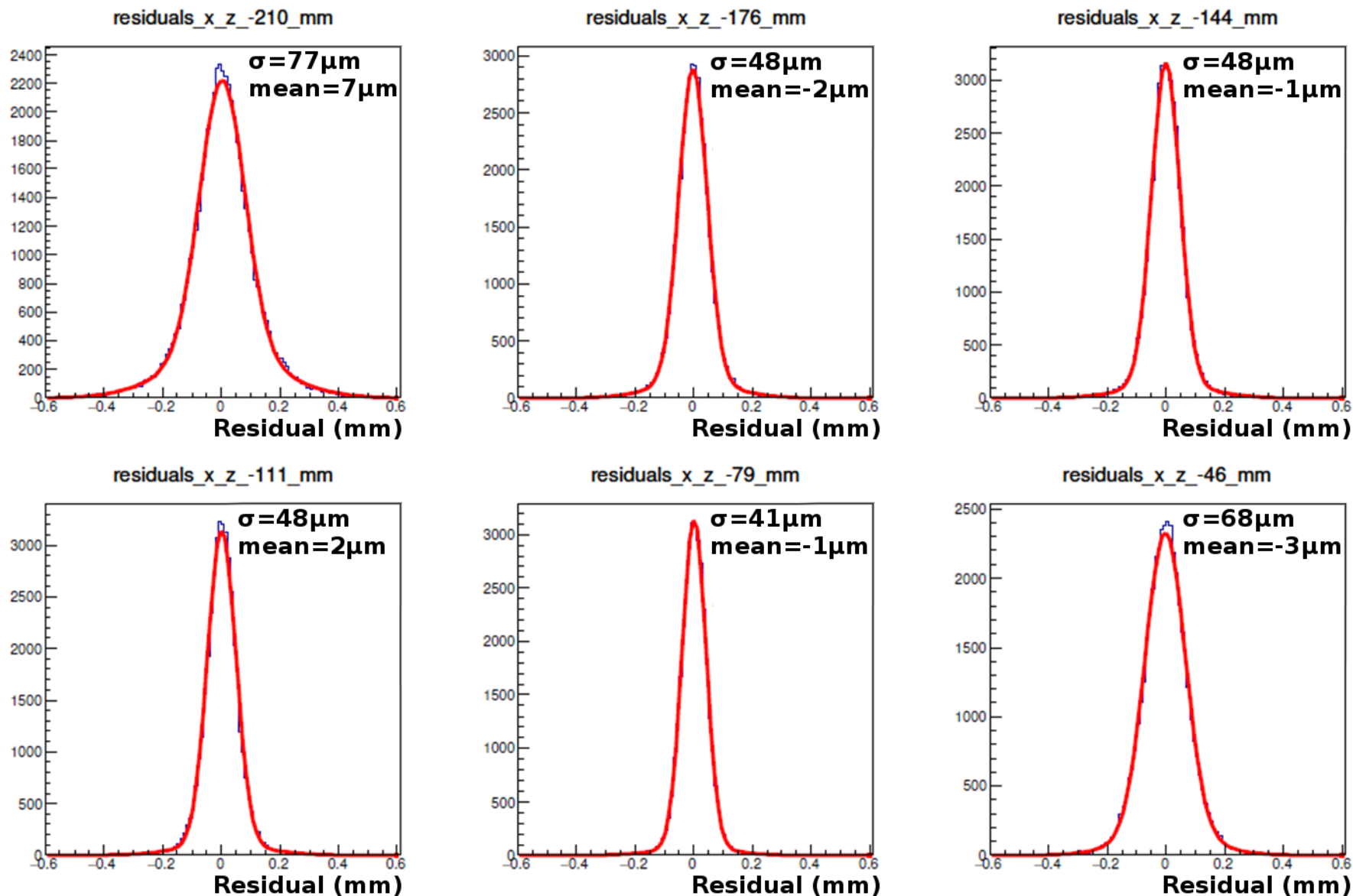
Test beam activity at CERN: ions



PRELIMINARY



STK resolution after alignment





The launch: Dec 17th 2015, 0:12 UTC

Jiuquan Satellite Launch Center
Gobi desert

Orbit: sun synchronous , 500km

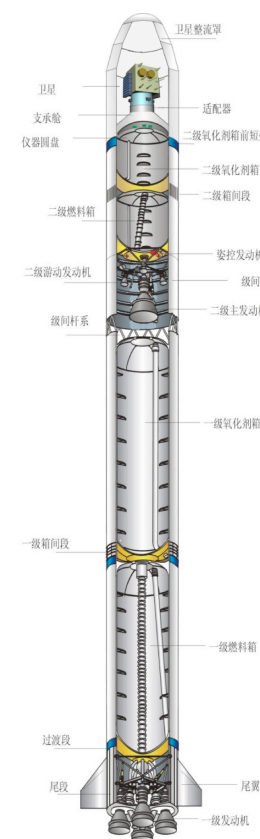
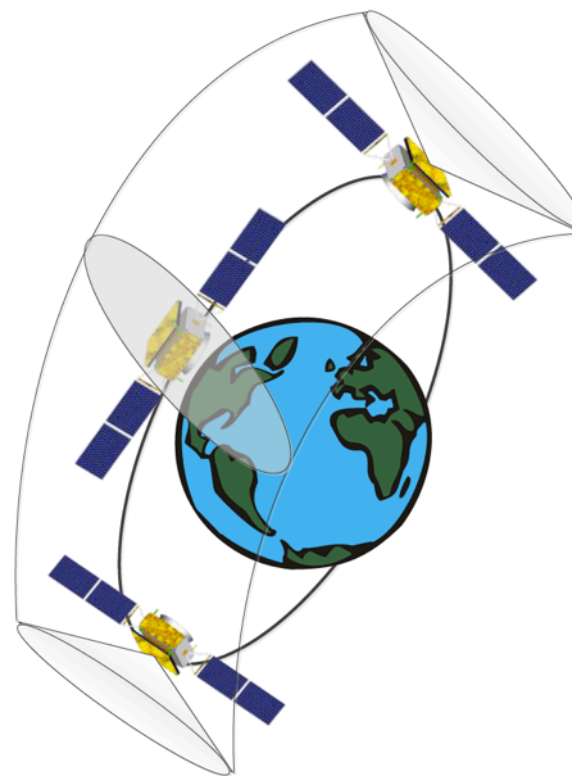
DAMPE → WUKONG



DAMPE Satellite

- **Launch: December 17th 2015, CZ-2D rocket**
 - **Total weight ~1900 kg, power consumption ~640 W**
 - **Scientific payload ~1300 kg, ~400 W**
 - **Lifetime > 3 year**

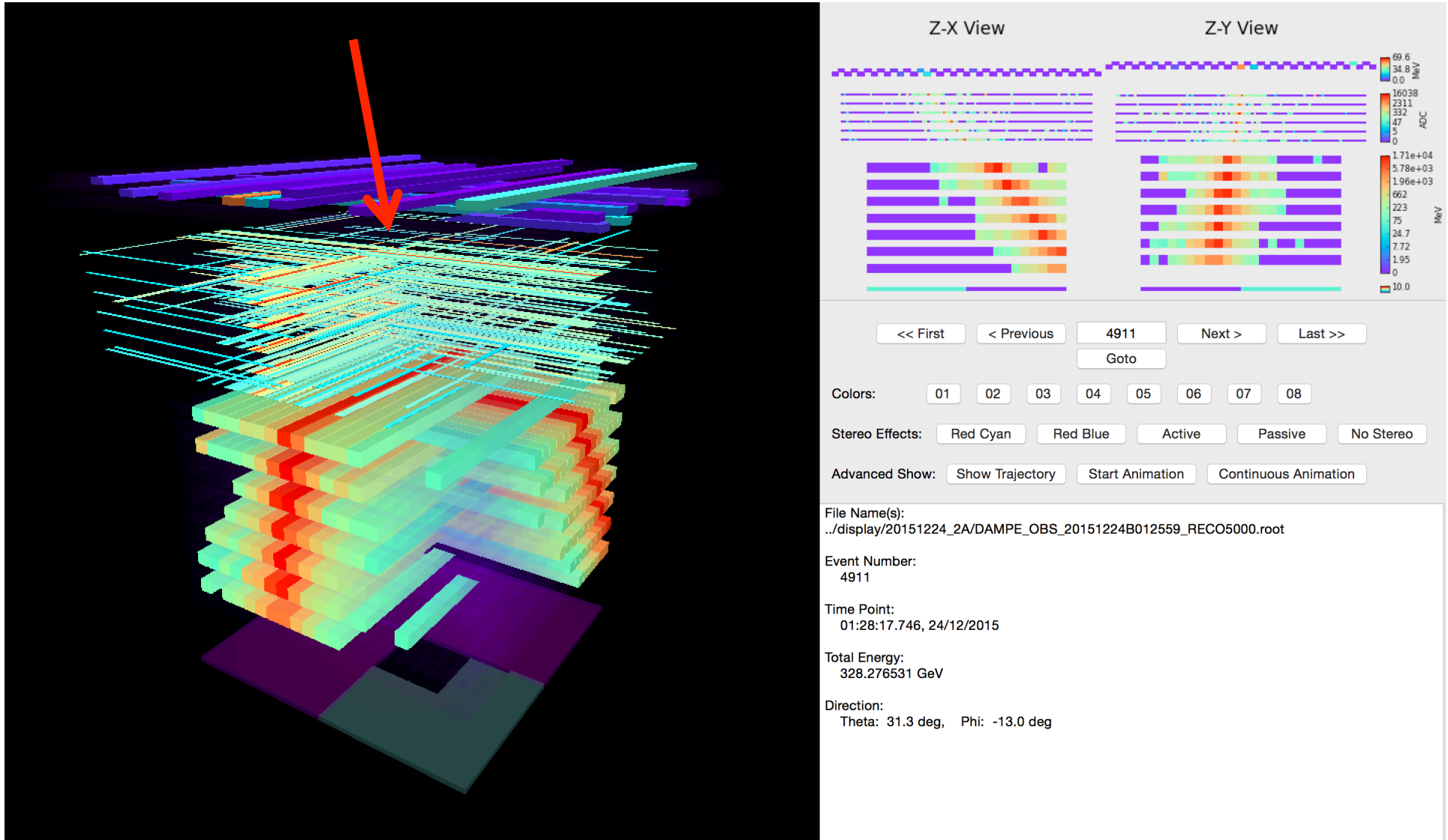
- **Altitude: 500 km**
- **Inclination: 97.4065°**
- **Period: 95 minutes**
- **Orbit: sun-synchronous**
- **12 GB/day downlink**



Dec 24th 2015: HV on



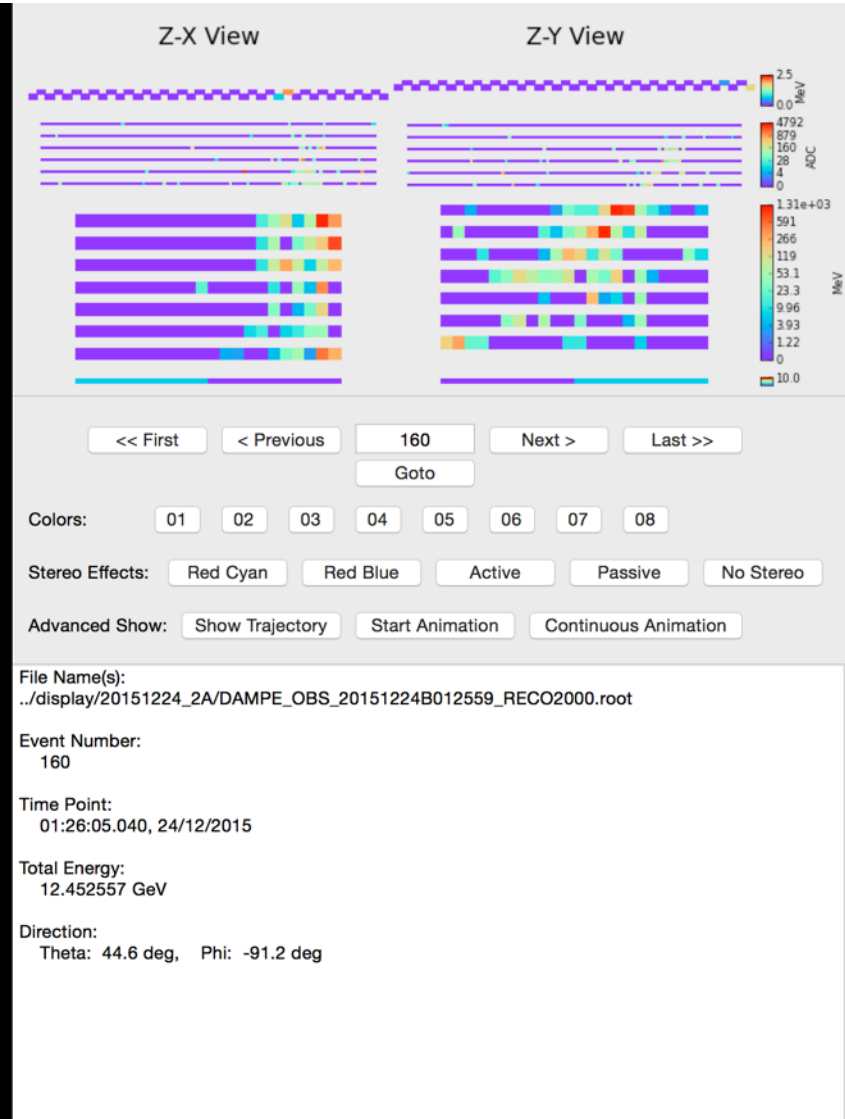
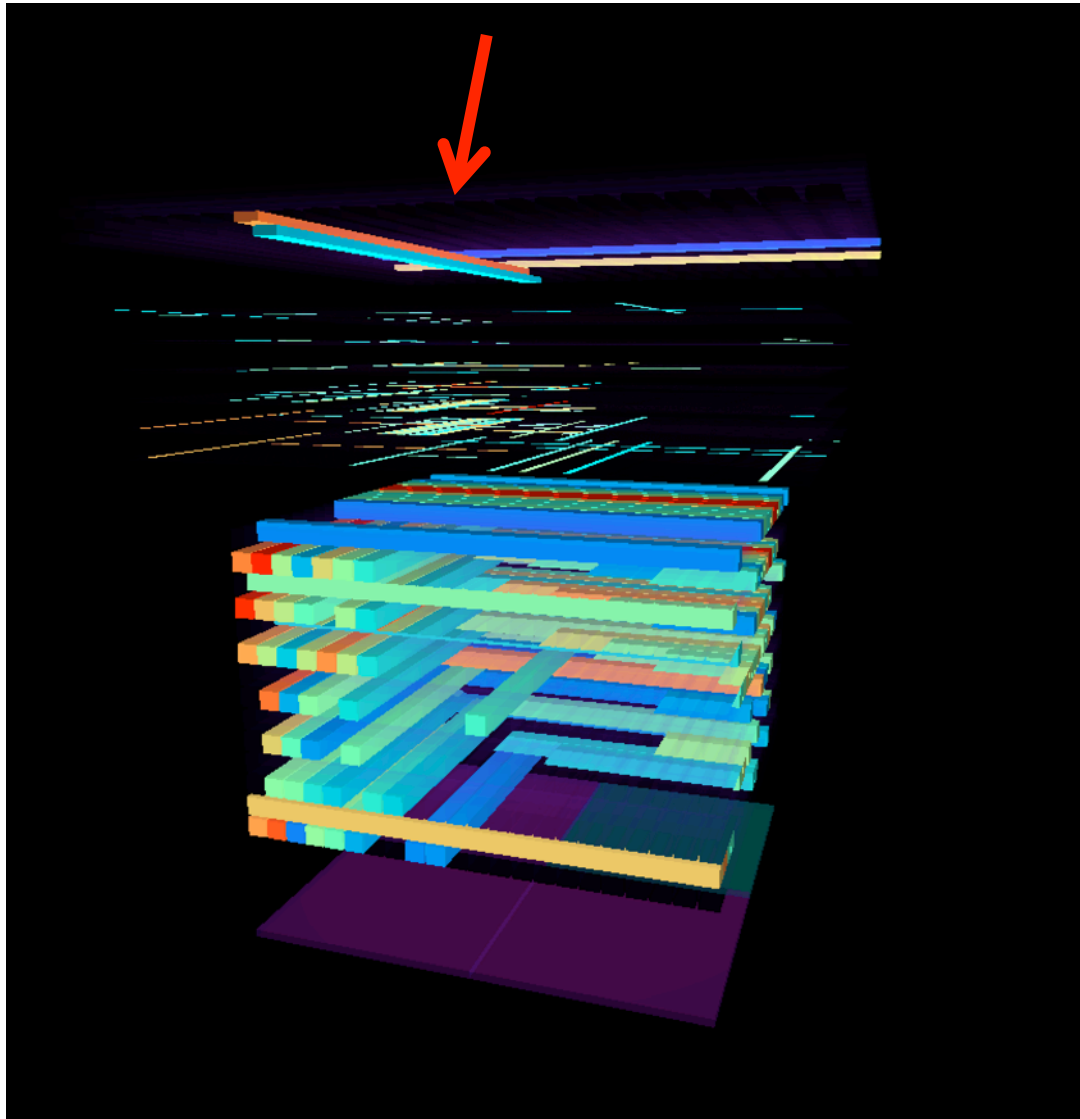
330 GeV electron



Dec 24th 2015: HV on



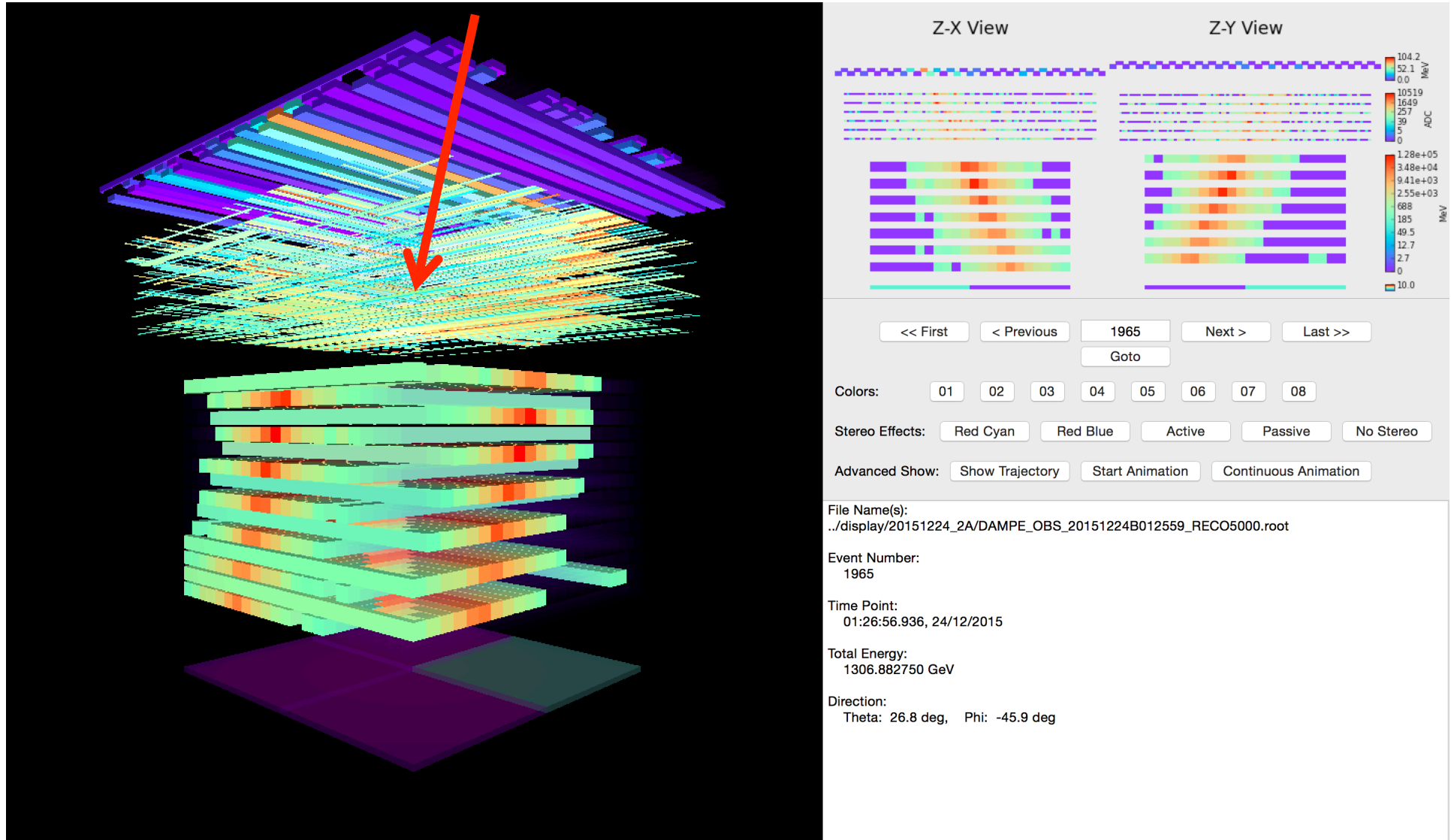
12 GeV proton



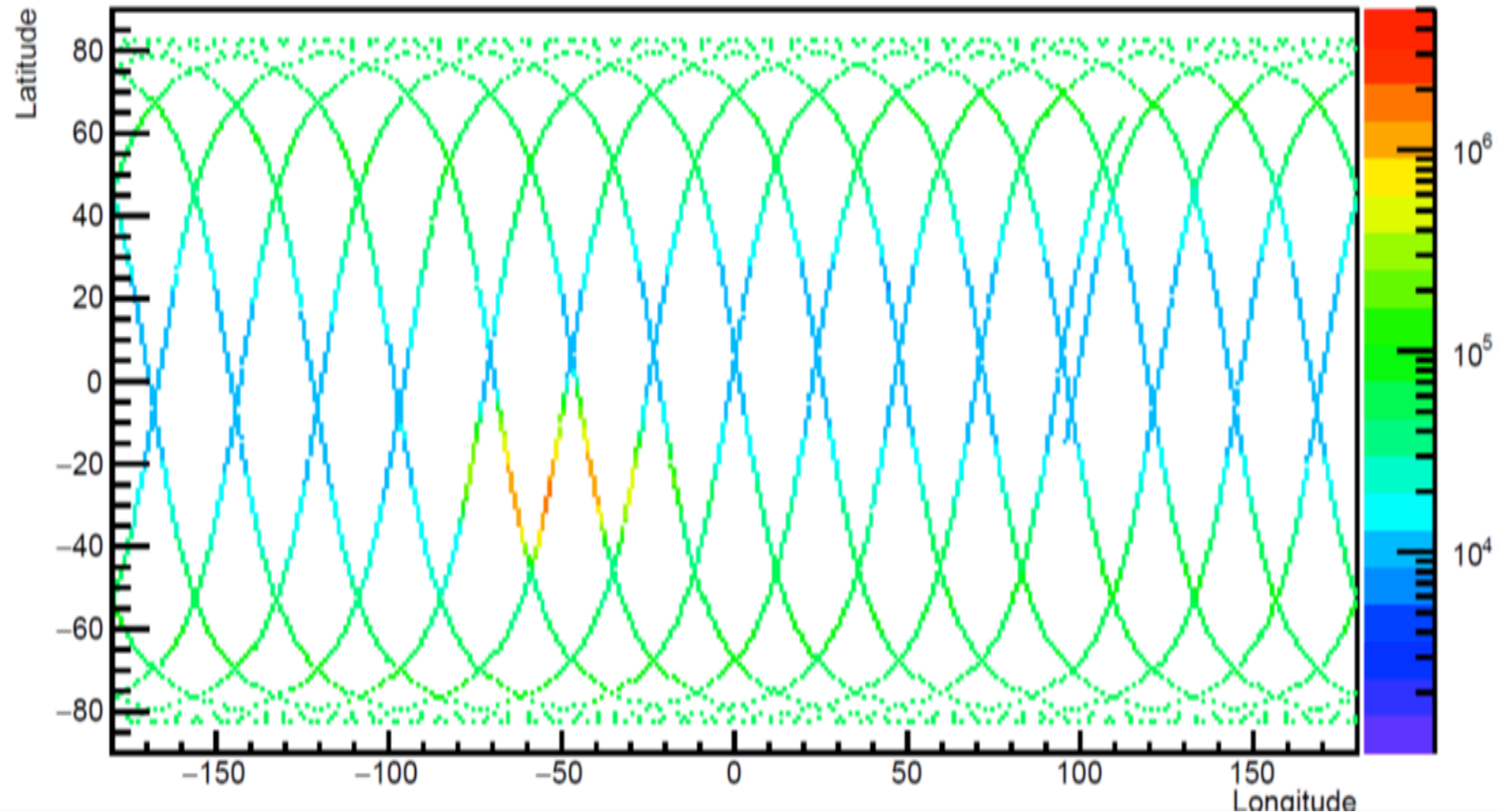
Dec 24th 2015: HV on



1.3 TeV carbon

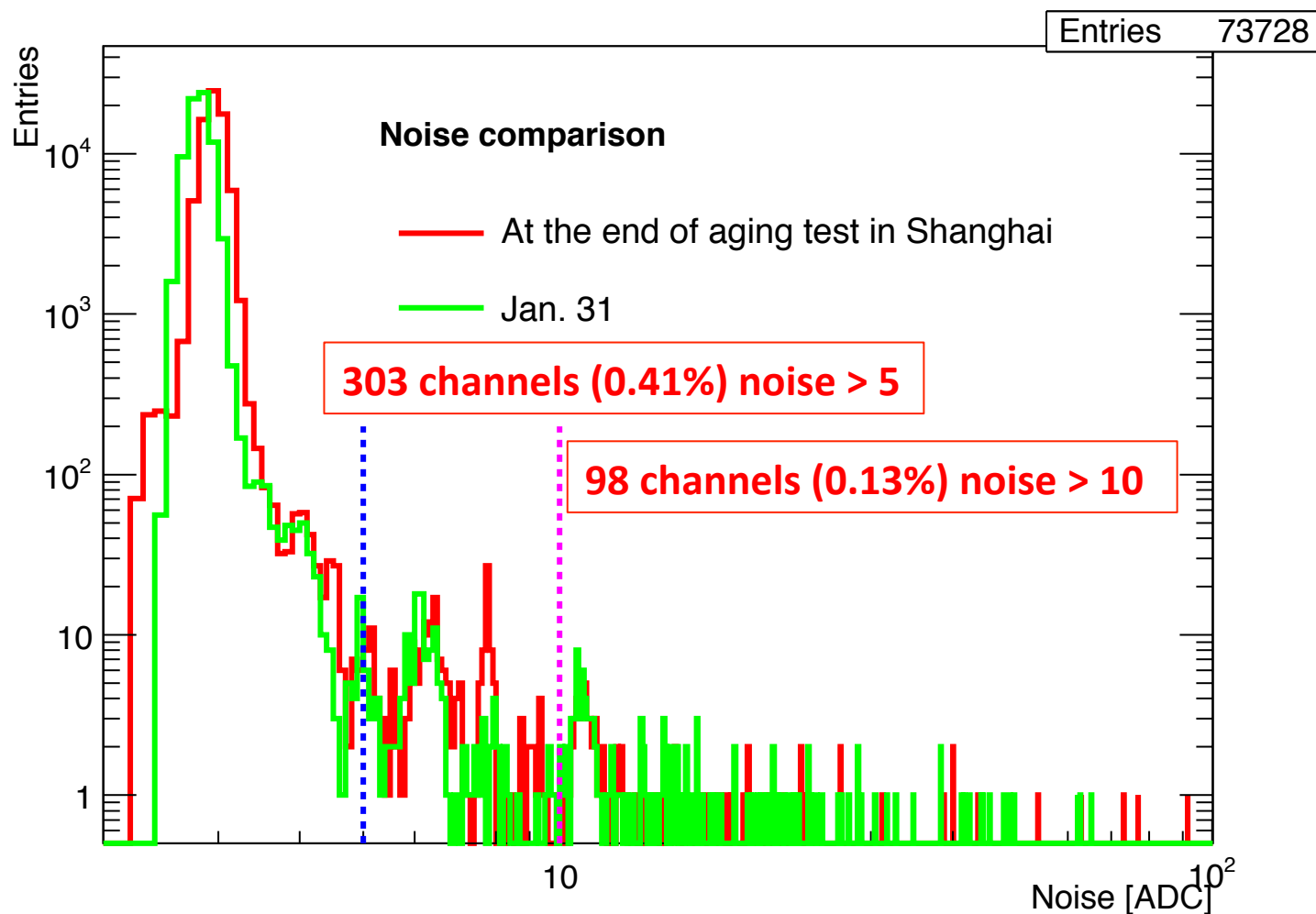


Trigger rate in orbit



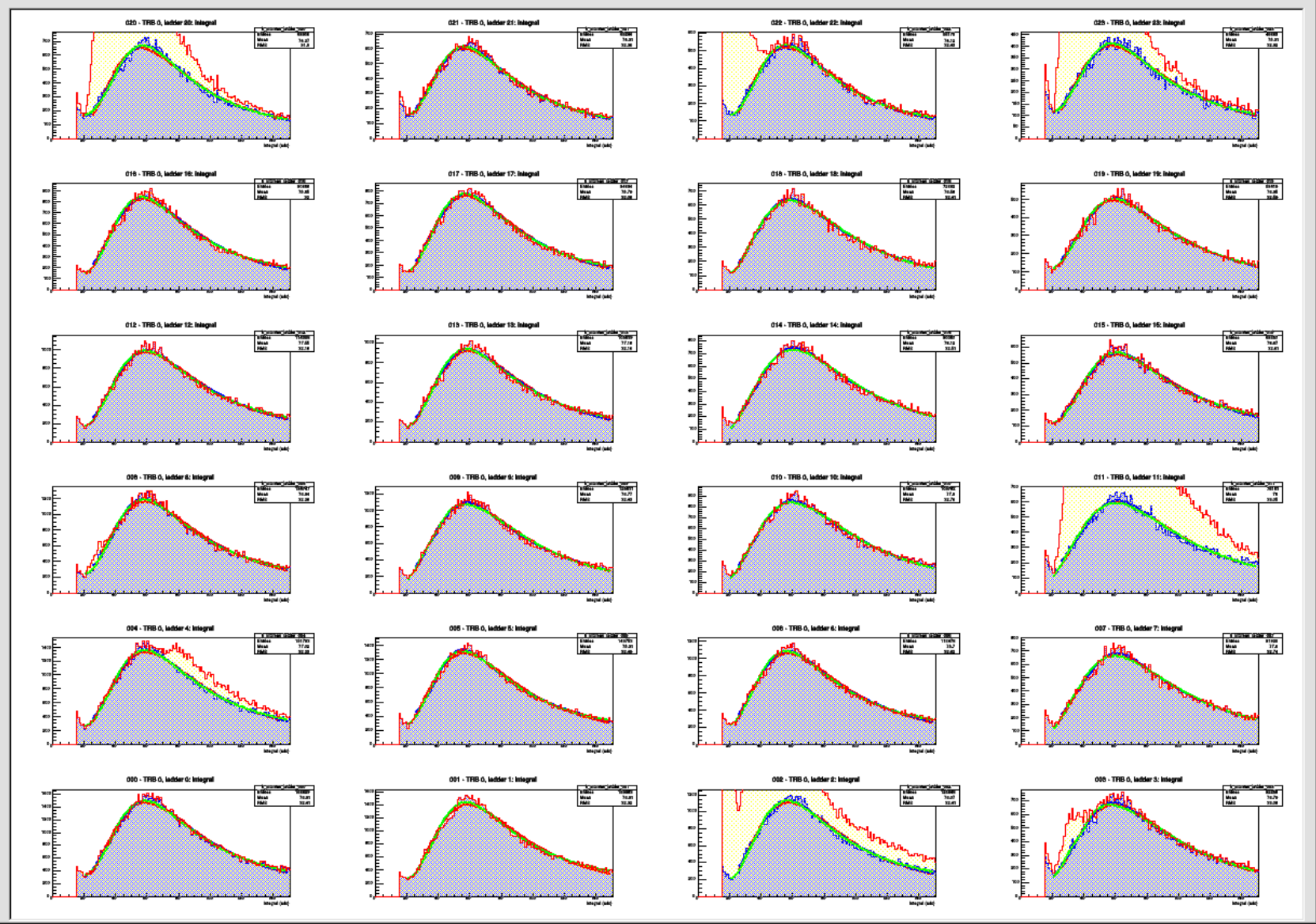
~50 Hz average trigger rate
→ 100GB/day on ground (about 4 M events)

STK noise behavior

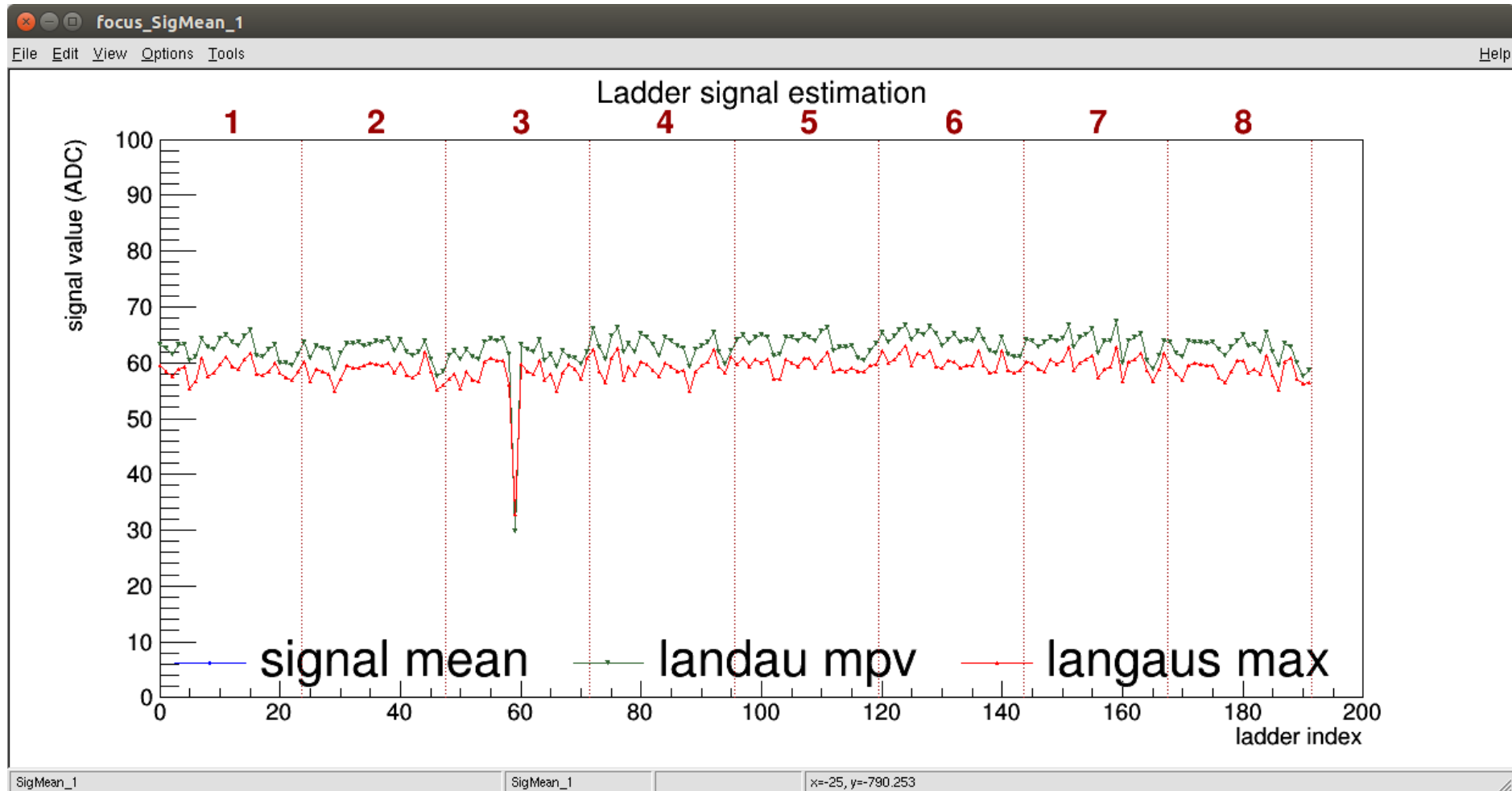


- Noise of the bulk significantly lower than on-ground

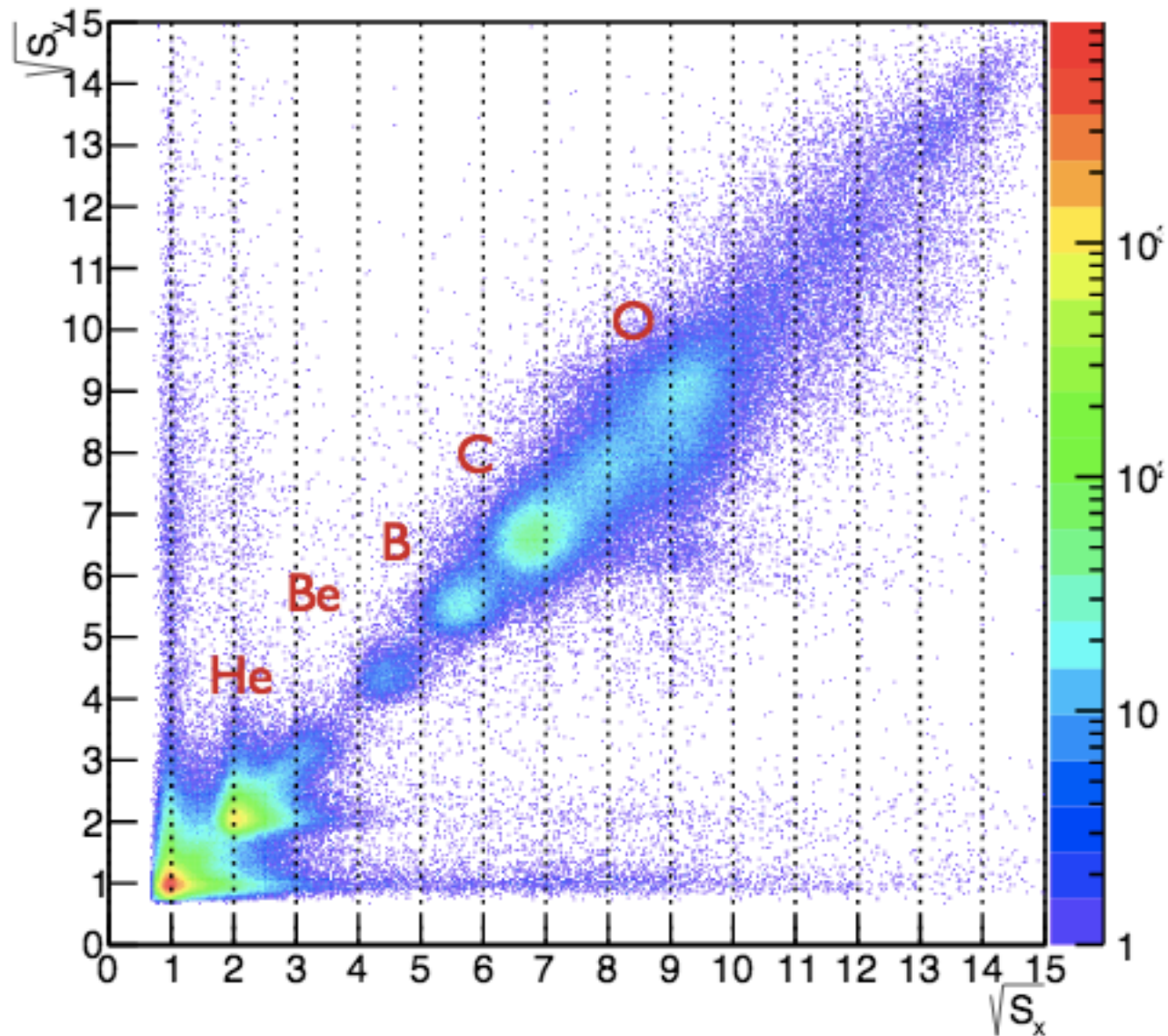
Cluster integral distributions



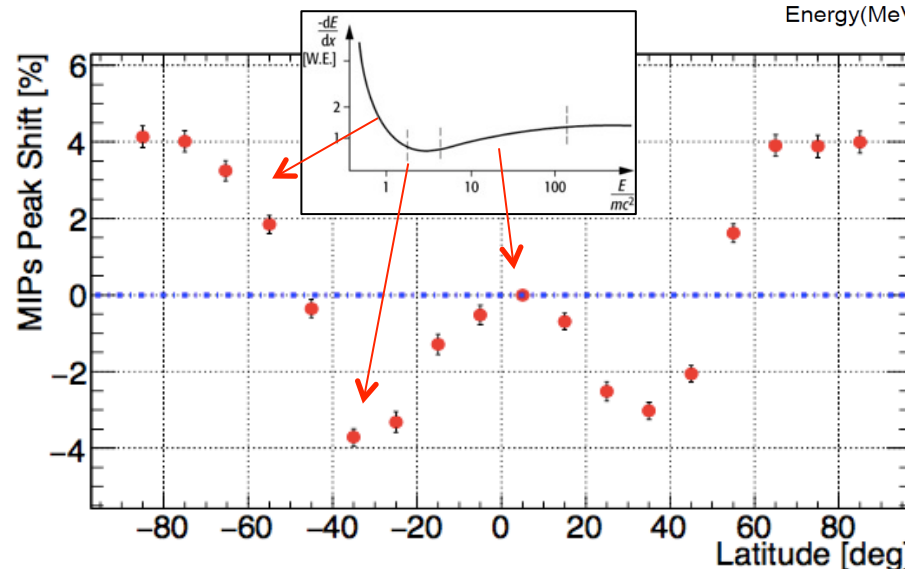
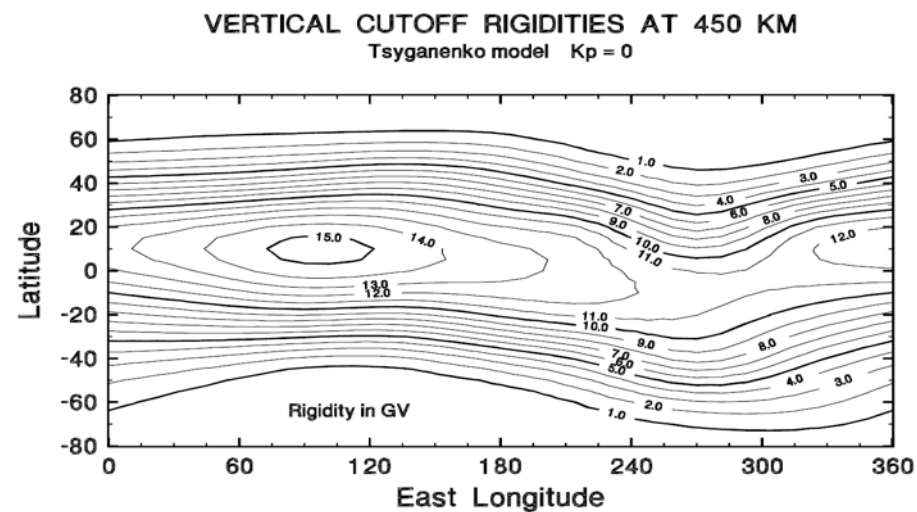
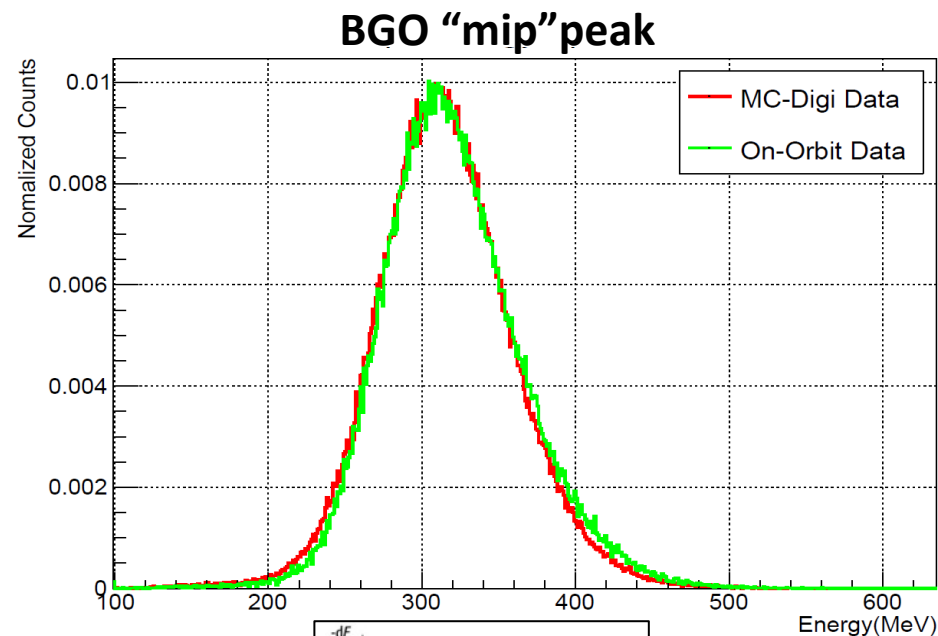
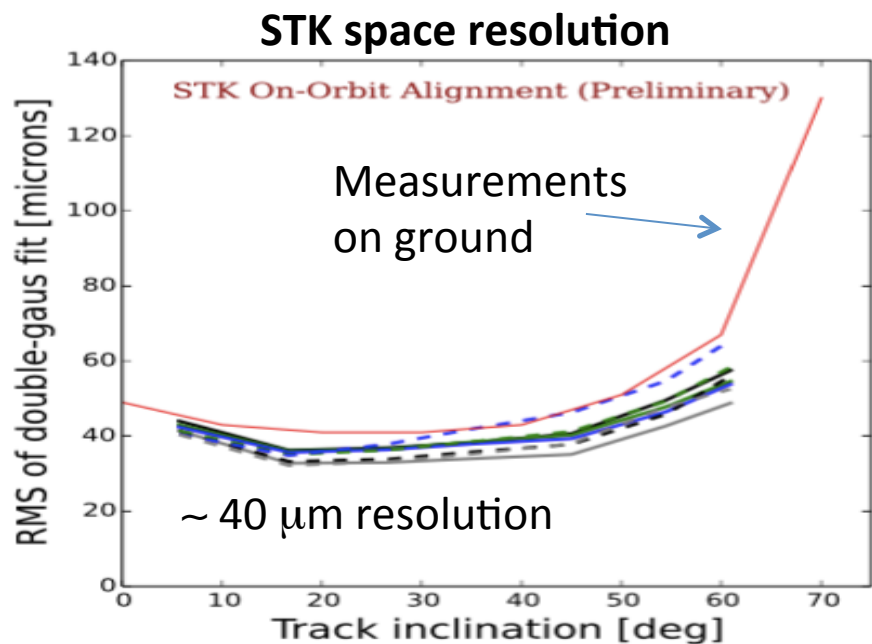
STK signal on all ladders



STK preliminary charge ID



More on-orbit performance plot

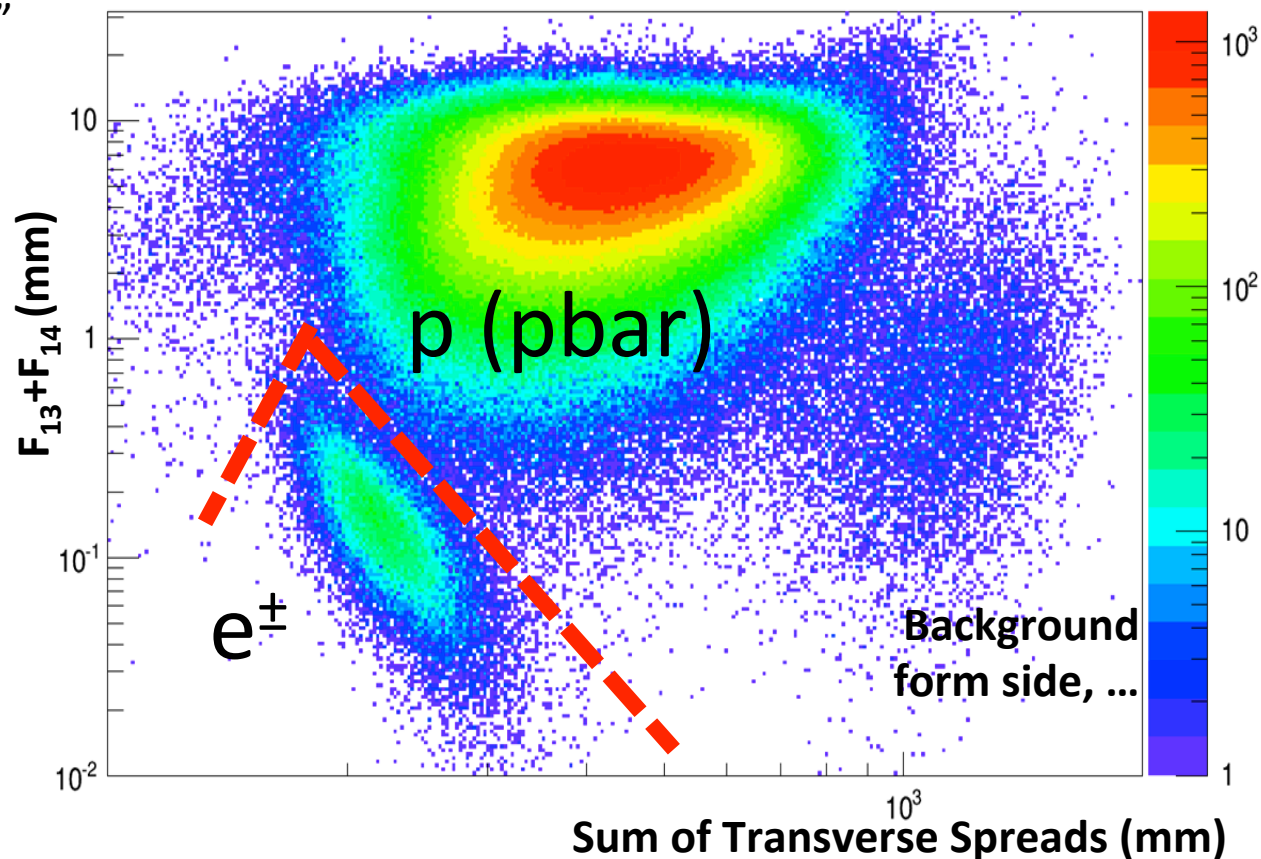
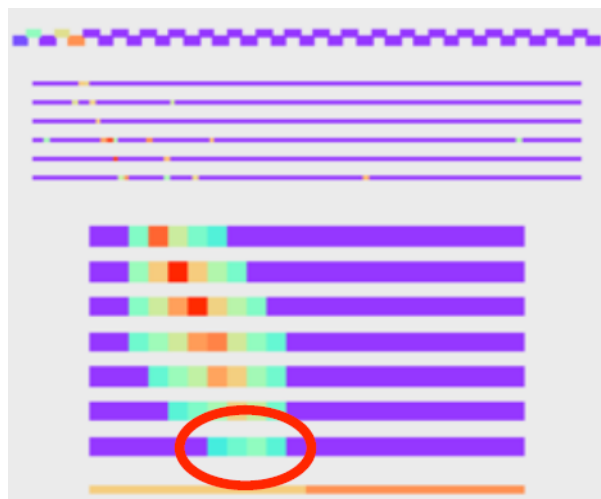


Electron identification

One possible “shape parameter”

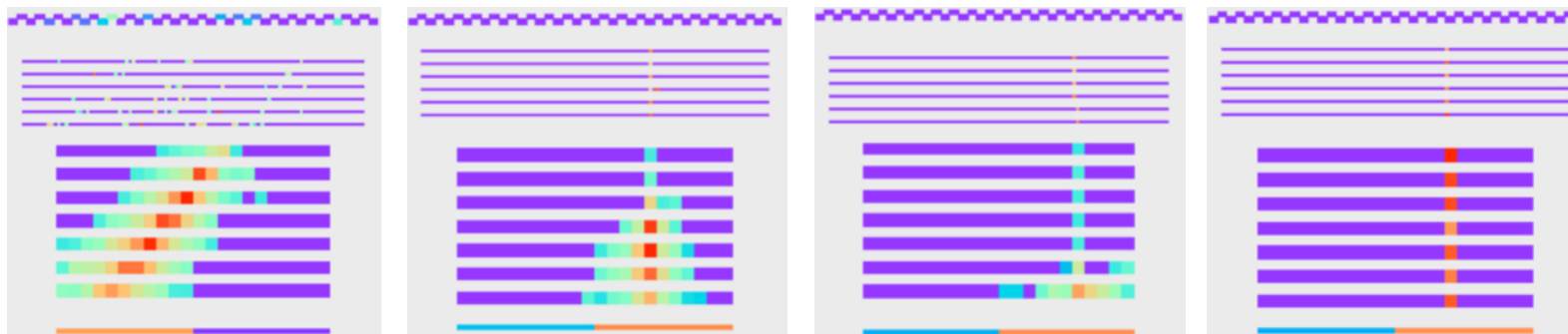
$$F_i = Spread_i \times \frac{E_i}{E_{tot}}$$

Rejection power $> 10^5$



Electrons and positrons

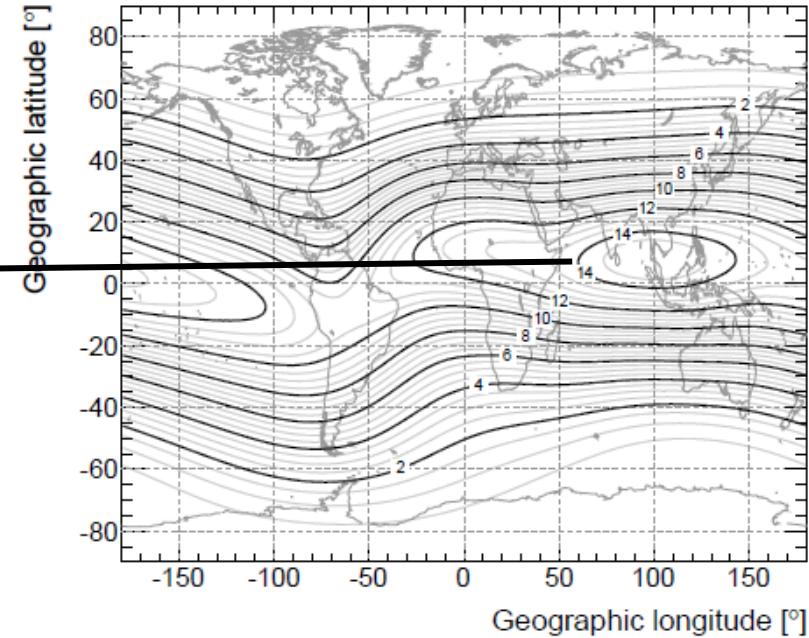
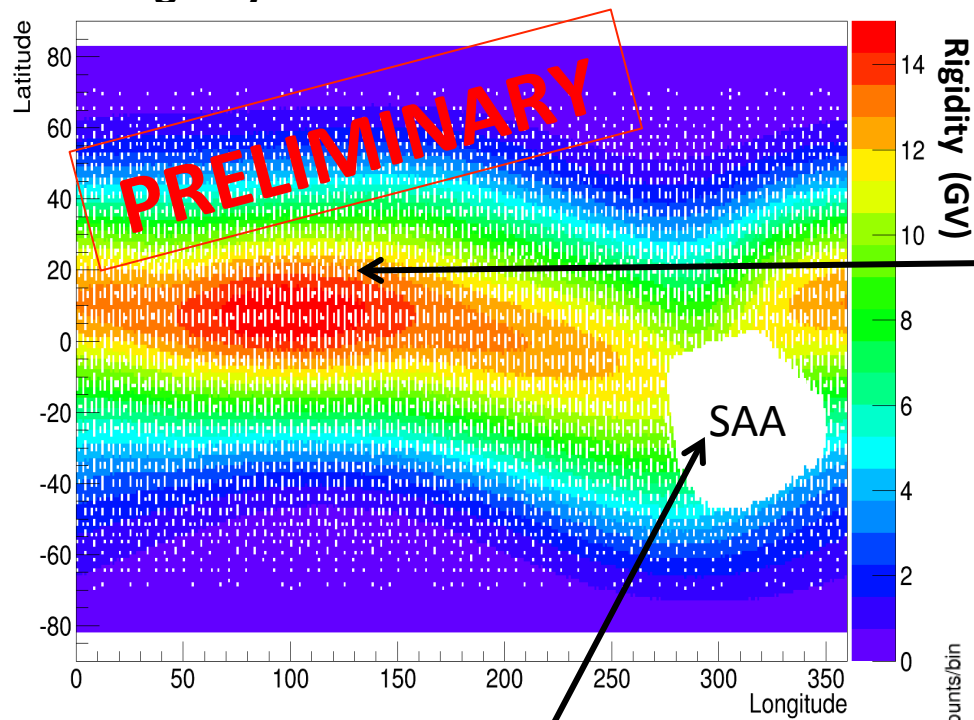
Protons and nuclei



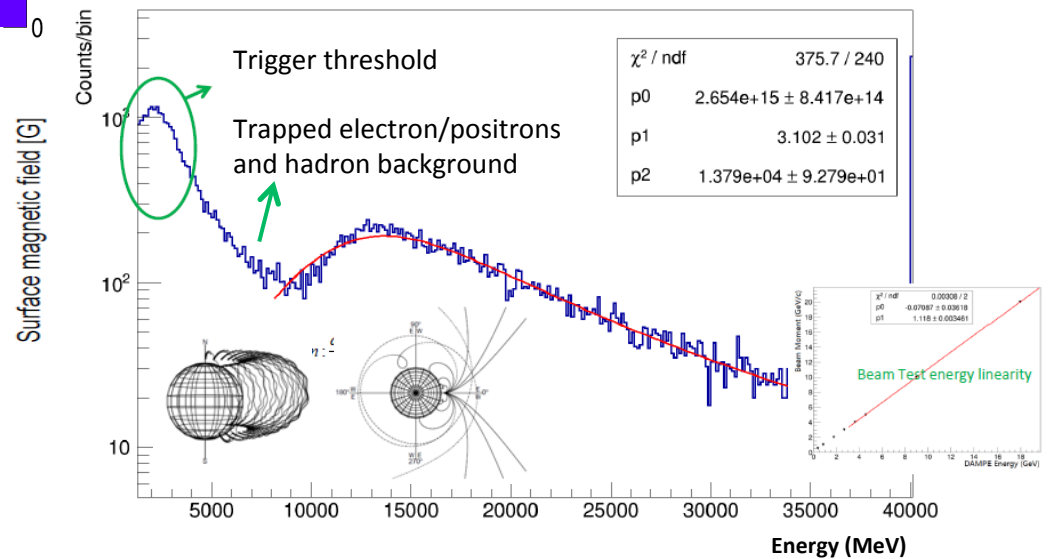
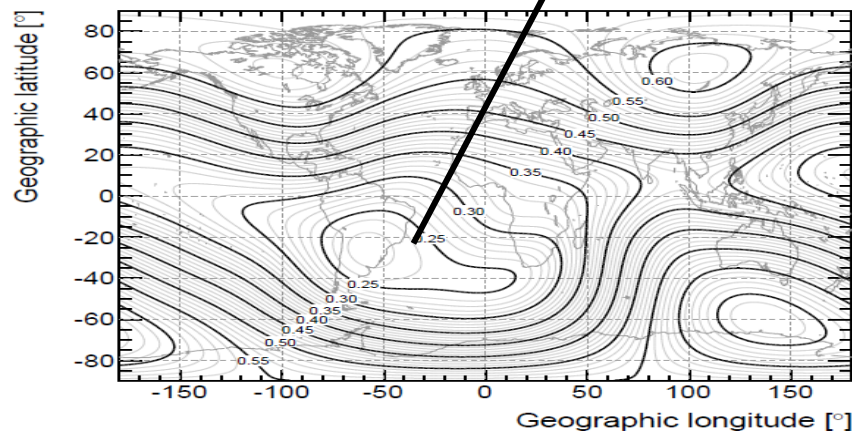
On-orbit energy calibration



e^\pm rigidity cutoff with 1 month statistics

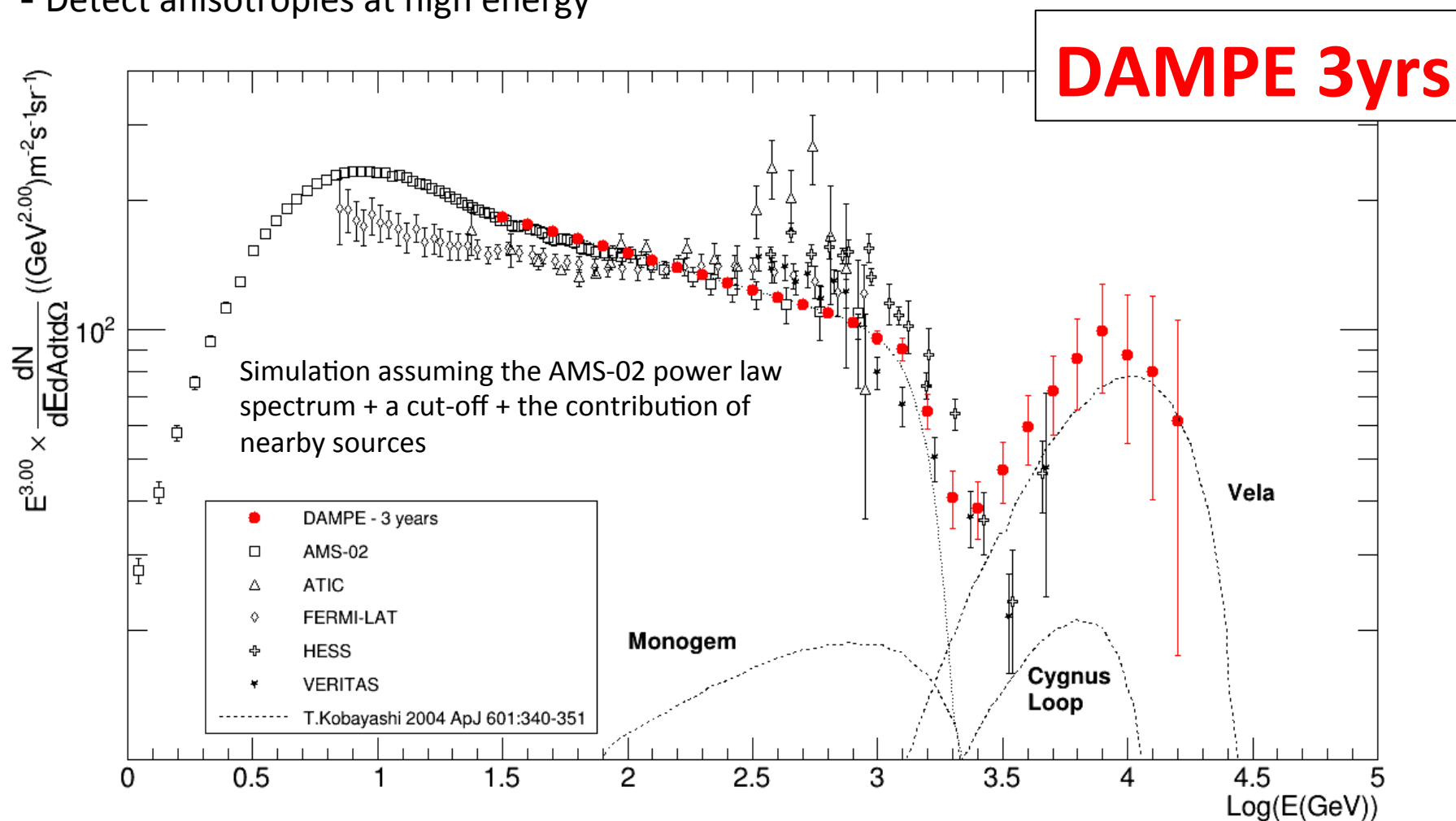


Vertical rigidity cutoff @ $h = 500$ km [GV]

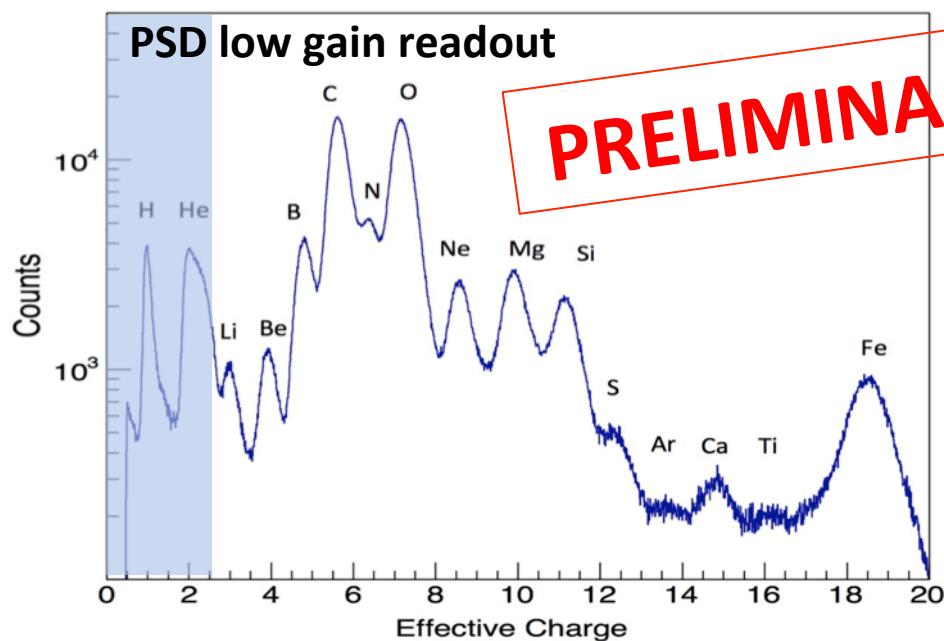
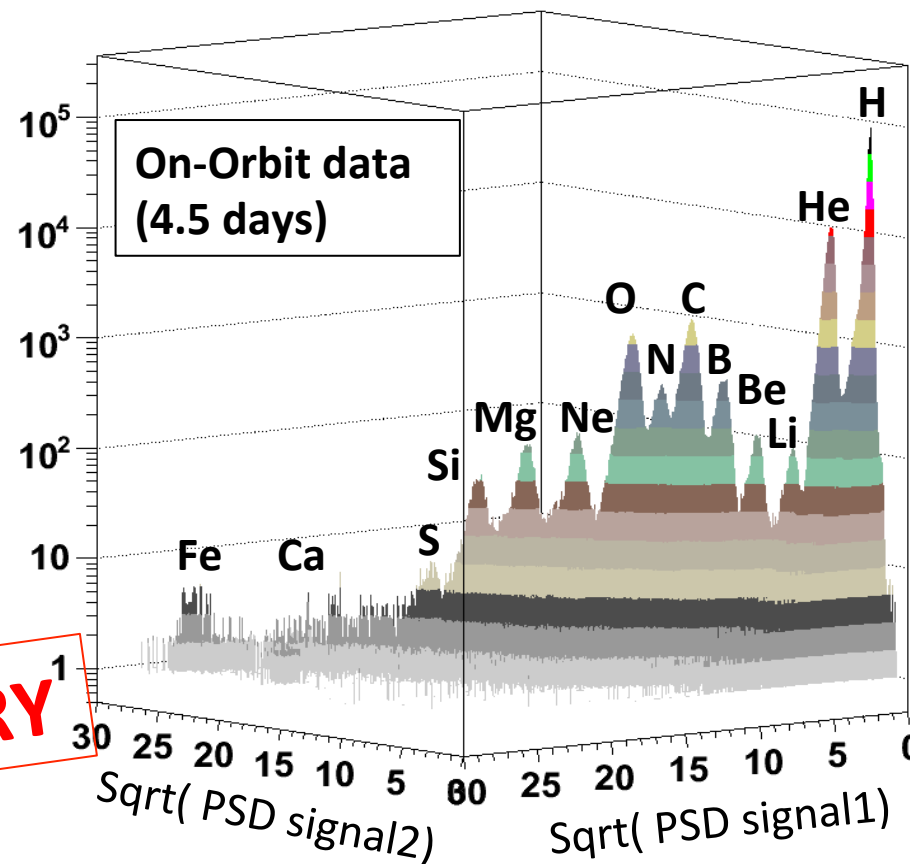
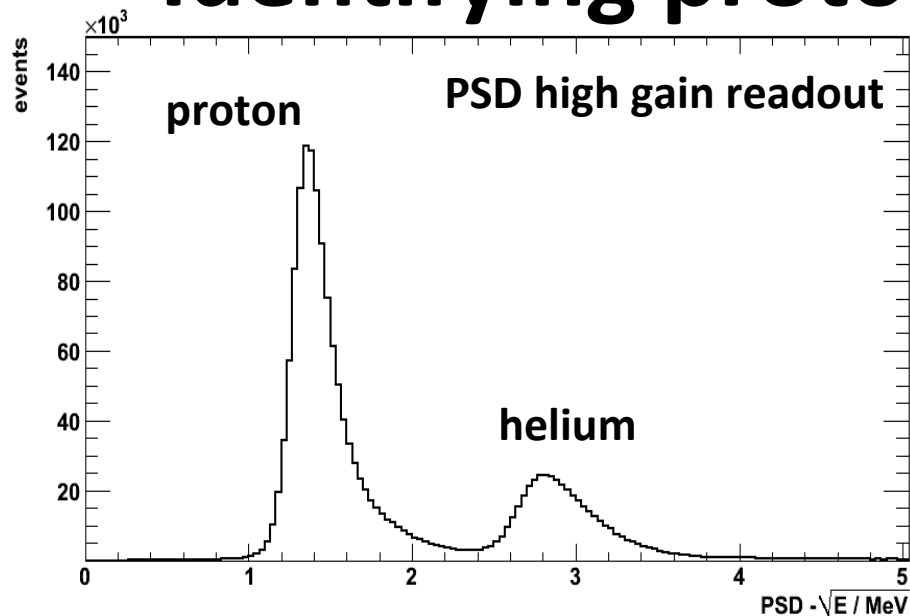


All-electron spectrum

- Measure the all-electron flux up to about 10TeV
- Measure with high accuracy the sub-TeV region and the possible cut-off around one TeV
- Detect structures in the spectrum due to nearby sources and/or DM induced excesses
- Detect anisotropies at high energy



Identifying protons and nuclei

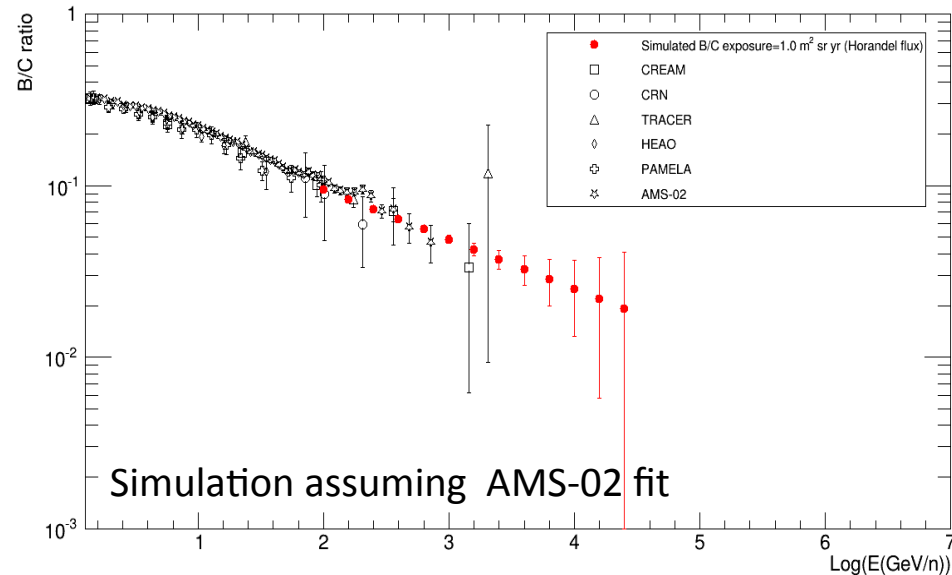
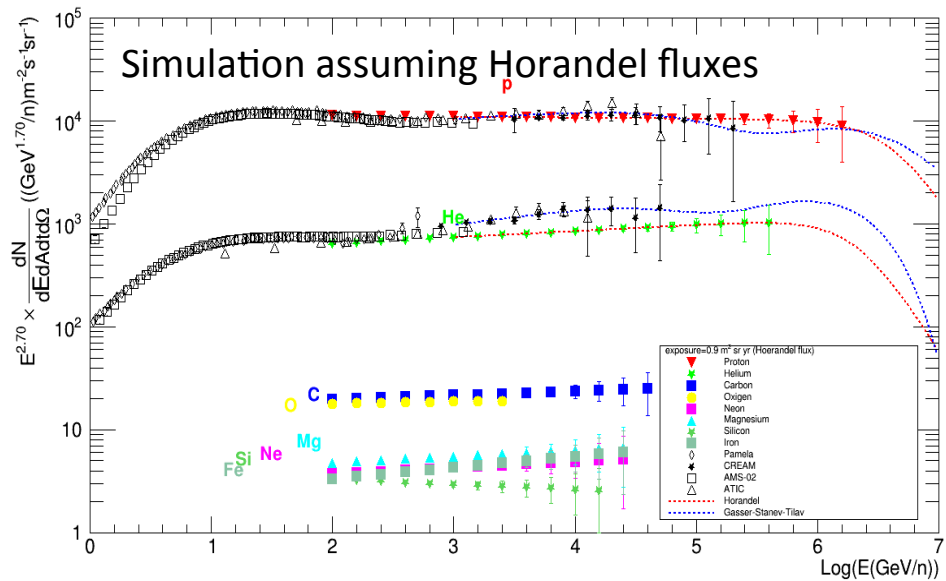
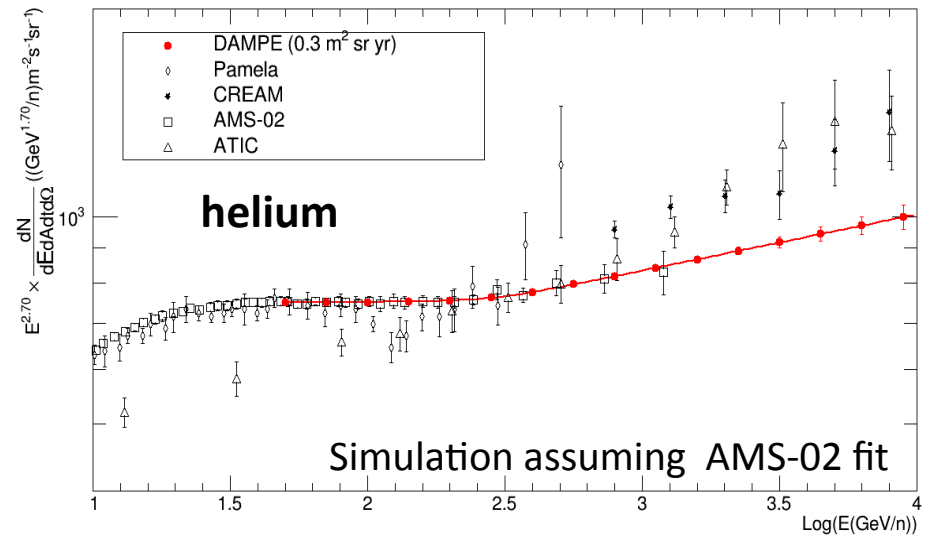
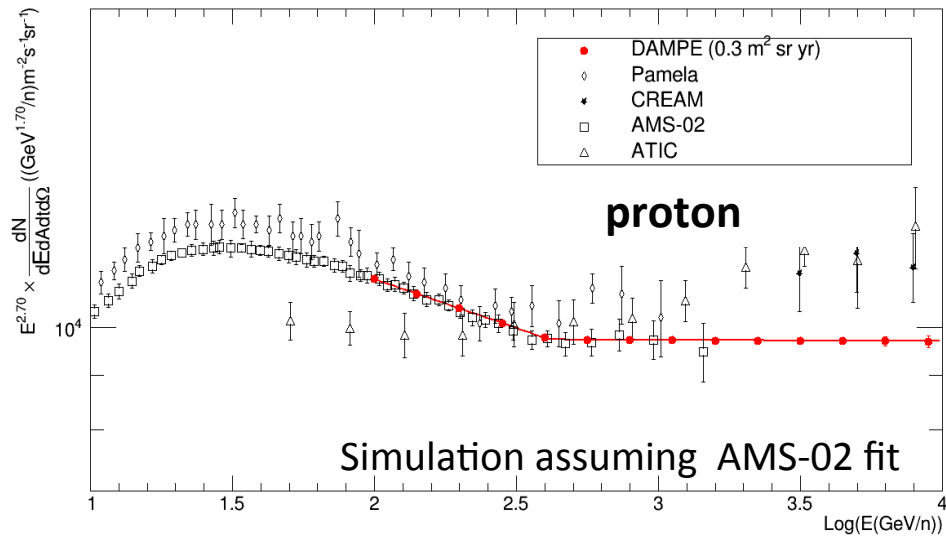


PRELIMINARY

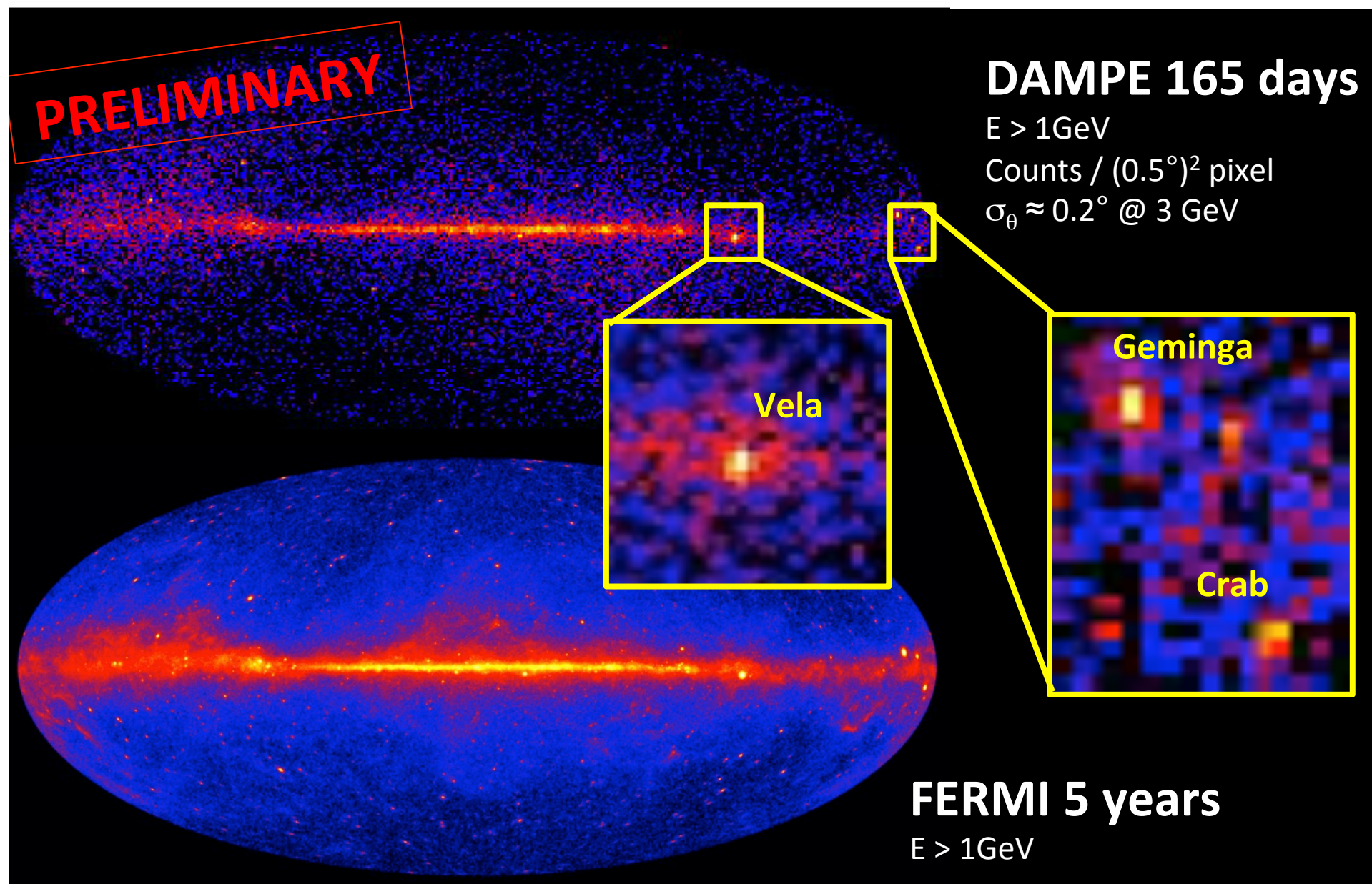
Charge resolution $\sim 0.2-0.3 e$ depending on Z

Charge measurement also given by STK and (with lower precision) by the BGO bars

Protons and nuclei: DAMPE 3years



Photons



Summary

The detector

- Large geometric factor instrument ($0.3 \text{ m}^2 \text{ sr}$ for p and nuclei)
- Precision Si-W tracker ($40 \mu\text{m}$, 0.2°)
- Thick calorimeter ($32 X_0$, σ_E/E better than 1% above 50 GeV for e/γ , $\sim 35\%$ for hadrons)
- “Mutiple” charge measurements (0.2-0.3 e resolution)
- e/p rejection power $> 10^5$ (topology alone, plus neutron detector)

Launch and performances

- Succesfull launch on Dec 17, 2015
- On orbit operation steady and with high efficiencies
- Absolute energy calibration by using the geomagnetic cut-off
- Absolute pointing cross check by use of the photon map

Stay tuned for physics results

